

# AVIATION WEEK

A MCGRAW-HILL PUBLICATION

MAY 19, 1952

50 CENTS



## Today, Guardian Angels Fly an ALBATROSS

Meet two guardian angels on the wing. Ahead, in the night, lives will be saved. Perhaps the survivors of a capsized sailboat on an inland lake, or a bomber crew downed on a hostile shore, or critically injured seamen on a tanker far at sea. Such are the missions performed in every corner of the globe by Air Rescue Crews of the Air Force. The big amphibian they fly is the versatile and rugged GRUMMAN ALBATROSS.

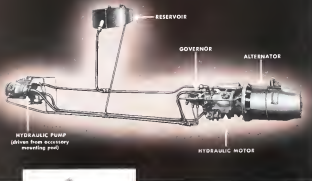
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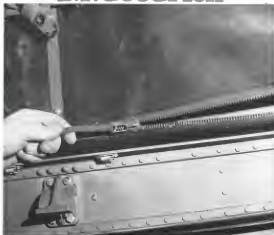


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# B.F. Goodrich



## It lets men in - keeps fumes out

**G**ASOLINE FUMES from the bomb bay fuel tanks in Lockheed's Neptune had to be kept completely away from equipment in the forward compartment. Yet the bomb bay couldn't be sealed off because a ready egress between the two compartments was needed. That called for an airtight curtain that would open in a jiffy.

Lockheed wondered if rubber could do the trick. They called in B. F. Goodrich. BFG engineers built a heavy mesh curtain to cover the entire width of the lower fuselage—made of fire-resistant fabric, covered with flame-retardant rubber. They made the curtain

straight-up completely removable—by means of a pressure-actuated zipper all the way around the edge.

That B. F. Goodrich-developed zipper has overlapping molded rubber lips. They provide a 100% seal against fumes. And they make it possible to open the fence curtain in no time at all. B. F. Goodrich fence curtains are now standard equipment on the Neptune.

Besides fence curtain applications, BFG pressure-actuated zippers are used for airplane doors, air ducts, streamer curtains, control surface seals. They are extremely flexible. They fit snugly around complex shapes—such as

squares, triangles and the like whose shapes won't seal. They are available in light duty and heavy duty sizes. They come right onto either fabric or metal. They save space and weight.

Other B. F. Goodrich products for aviation include tires, wheels and bushings, heated rubber, De leet-Airtron, Plastidur, adhesives, inflatable seals, fuel cells, E-waves, accessories. The B. F. Goodrich Company, American Division, Akron, Ohio.

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C. Inc., Auburn, N.Y.

# Aviation Week

Volume 54

May 19, 1952

Number 20

## Headline News

CAR Progress Jet Transport Study... 12  
AF Plans Test Stratospheric... 13  
Panavia Brothers T400... 13  
AP's Shoppers: YB-49 & YB-48... 14  
AF's Defense System... 15  
C-W Joint Asia New Industry Policy... 16  
Truman... 17  
Aerospace Council (SAC) Introduced... 17  
Salem to 1957 House... 18  
Ruler Air Travel Outlook Bright... 21  
Aerobics Soda Air Race Equipment... 22

## Aeronautical Engineering

U-480-D Study Yields Speed Data... 15

## Production

Old Jet Sparhawk Test Late... 26  
Designed/Designed Shown Planned... 40

## Departments

New Design... 41  
Aviation Calendar... 42  
Federation Knowledge... 43  
The's Where... 44  
Industry Observer... 45  
Threat & Long... 46

## Shortcuts

41,691 copies of this issue printed

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Editorial Office: 120 West 44th St., New York 36, N. Y. Phone Langmuir 4380, or  
Langmuir 4381. National Press Bldg., Washington 4, D. C. Phone National 3484.  
Communications: New York: 400 N. 10th St., Minneapolis 1, Minn.; Chicago 11, 220 N.  
Michigan Ave.; Cleveland 11, 10000 10th St.; Detroit 26, 10000 10th St.; Los Angeles  
11, 1111 W. 10th St.; San Francisco 4, 1000 10th St.; Boston 14, 1000 10th St. Our  
correspondents in more than 50 major cities.

Foreign: New York: London, Paris, Frankfurt, Tokyo, Bombay, Melbourne, Rio de  
Janeiro, Mexico City. Correspondents in more than 50 major cities.

Aviation Week is owned by Fafnir Associates, Inc., a subsidiary of Associated Press.

Robert F. Roper

PUBLISHER

R. W. Martin, Jr., Advertising Sales Manager; J. G. Johnson, Business Manager; Mary  
Kempson, Research and Marketing, Sales Representative; J. C. Anderson, New York;  
J. P. Johnson, Cleveland; J. J. Day, Chicago; W. E. Dunsen, St. Louis; R. P. Shan-  
non, Jr., Boston; James Cook, Dallas; R. C. Sheldahl, Atlanta; S. E. Dunsen, Jr.,  
San Francisco; C. F. McElroy, Los Angeles; W. S. Brown, Indianapolis. Other  
sales offices in Pittsburgh, Detroit, London.

May 19, 1952

AVIATION WEEK

Vol. 54, No. 20

AVIATION WEEK is published weekly (except for two issues combined) by Fafnir Associates, Inc., 120 West 44th St., New York 36, N. Y. Phone Langmuir 4380, or Langmuir 4381. National Press Bldg., Washington 4, D. C. Phone National 3484. Communications: New York: 400 N. 10th St., Minneapolis 1, Minn.; Chicago 11, 220 N. Michigan Ave.; Cleveland 11, 10000 10th St.; Detroit 26, 10000 10th St.; Los Angeles 11, 1111 W. 10th St.; San Francisco 4, 1000 10th St.; Boston 14, 1000 10th St. Our correspondents in more than 50 major cities.

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## Articles

ATA Seeks Status With Low Profile... 27

## Equipment

Full Oil Tanks Boost Fuel Systems... 47  
Tank Contents Target Fuel... 48

## Finearts

Non-Artistic Enterprise Studied... 47

## Air Transport

Albion Express Tanker Program... 49  
Continental Airline Plans New... 50  
Western Airs CAB Ending on Airlines... 51

## Editorial

Class Expansion... 49  
Engineer Becomes Engineer... 49

## Departments

Production on Drawing... 49  
New American Products... 51  
Also on the Market... 51  
Letters... 52  
What's New... 52  
USAF Comments... 52

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non, Jr., Boston; James Cook, Dallas; R. C. Sheldahl, Atlanta; S. E. Dunsen, Jr.,  
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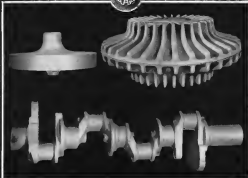
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## NEWS DIGEST

### Domestic

**Indoportment** of the 1952 National Air Races until 1953 was announced last week by Ben V. Franklin, general manager. However, Aero Club of Michigan which represented last year's National Air Races at Warren Motor Airport, Detroit, will hold its annual air fair at the same time and to coincide in conjunction with Air Force Armory convention and National Aero-nautics Association convention. Both will meet in Detroit over the Labor Day weekend.

**International Air Transport Association** last week called an emergency session in Paris as a result of the serious gas shortage caused by the U. S. oil strike. A reported one third out of service was being demanded for most major European lines. Meanwhile at Aviation Week press time, Major Sublimation Board was undergoing a pay sitting through for the 98,800 striking of workers.

**AIA Board of Governors** took their annual meeting at Wilkesboro, Va., May 17-18, will discuss joint industry government conference on aviation production problems. The session will be closed because of their elevated where.

**Henry W. Fowler** has been announced by President Truman to succeed Mervyn H. Hines as Defense Production Administration. Fowler, is presently managing June 1, had been of David C. Fowler, Jr. who's former post as Defense Secretary, has been put in Defense Secretary.

**New international airlines** reveal for Category II aircraft was established May 1. In New Council through a Paper Race Last August-New York, starting (246) miles. Council under the law is 24 for 58 miles. Previous week was 2,156 miles in 1950 in the late John F. May.

**Flight Safety Foundation** has named Theodore Wright, J. Carlton Ward, Jr. and Kenneth M. Cleveland to its governing board.

**Paul of New York Authority** has awarded a \$3,131,857 contract to Van Construction Co., N. Y., for construction of new 54-million passenger terminal building at Newark Airport. It is scheduled for completion in 1957.

**Paul Public, United Aircraft Corp.**, director of public relations, planned to



**THE WREATHS CASE** OFF USAF's two new heavyweight bombers last week, when Air Force disclosed general extent details of the Wright-Rising XR-52 and Corvus XR-56. The photo, just released, shows the

assembly-depot town of the XR-52 after it was rolled out of Boeing's Seattle plant at about 3 a.m., Nov. 20, 1951. Other photos of the XR-52 and XR-56 appear on pp. 14-15 of this issue.

return to his last Hartford, Conn. home last week following an accidental operation at Hartford Hospital. After several months' convalescence he is to undergo further surgery.

**Martin XR-51** flight test bomber crashed and landed at Edwards AFB, Calif., May 9 while performing a low-level flight maneuver at moderate speed. The XR-51 pilot was killed. Crash, but not been determined. The only other XR-51 built is being returned to the Martin plant for repairs following an "accidental" landing at Wright-Patterson AFB, Ohio.

**Gen. Nathan F. Twining** is acting USAF Chief of Staff while Gen. Hoyt H. Vandenberg commences following an "exploratory" abdominal operation at Doctors Hospital, Washington.

### Financial

**Solar Aircraft Co.**, San Diego, reports successful sales of over \$1 million for the first year ended Apr. 18, double the sales for the previous fiscal year.

**G. M. Giovanni & Co., Franklin, Calif.**, aircraft and engine instrument order, had sales of over \$1 million for the first quarter of 1952. Sales for entire 1951 were \$2,571,129. A type graphed error was made last week in printing these sales figures.

**Bell Aircraft Corp., Bufile, N. Y.**, sales and revenue of \$14,030,494 for the first quarter ended Mar. 20, with net profit after taxes being \$267,751. Net profit was nearly identical to the 1951 figure in spite of nearly \$1-million increase in quarterly sales.

**Trans World Airlines** reports a net loss of \$412,823 for the first quarter of 1952 despite more than 53 million income, in gross revenues over same period last year.

**Tenneco Aircraft Corp., Dallas**, had net earnings of \$185,800 on gross sales of \$18,054,000 for the quarter ended Mar. 31. Tenneco's backlog is of Mar. 31 was \$195.5 million.

**Lockheed Aircraft Co., Burbank, Calif.**, has declared a 10-cent per share of common stock payable June 13 to holders of record on May 21.

**Chrysler & Southern Air Lines** has declared a dividend of 15 cents per share of common stock payable June 16 to holders of record June 2.

### International

**Indian navy** has ordered ten Short Sealand amphibians for reconnaissance duties, with first delivery expected to be made toward the end of this summer.

**Trans Australia Airlines** has ordered five Vickers Viscount transport turboprops. Delivery dates are not yet known.

**Blaker Brothers "super priority" fighter for RAF** is also to be built by Glaser Aircraft Co.

**KLM DC-6** which crashed in a Friesland, Germany, suburb on Mar. 22 taking 45 of 47 aboard encountered undershoot pump reversal, according to the best information that is now available.





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## WHO'S WHERE

### In the Front Office

John W. Hutchinson is being sound and? in very product sales and traffic of Longman, American Airlines, as well as the recent success in obtaining a large number of contracts, according to L.A.A. president Woodrow F. Lush. Hutchinson has ten years as director and manager in Chicago for Northern Airlines and also served NAA in Washington representative and sales manager in that area for a period of three years.

### Changes

Lawrence J. Walden has been appointed as national public relations manager of Lockheed Aircraft Division. He comes from Syracuse University where he was director of public relations.

W. F. Moore has been designated manager aviation department of General Electric Co. Pacific Northwest District with headquarters in Seattle. C. E. Flinn who formerly held this position has been transferred to Giles Aircraft Nuclear Propulsion Project with headquarters located in Oakland, Ohio.

Don Downer, western editor for Bureau magazine and aviation writer-photographer for the Freedom News, has joined Kinkaid American Corp. Functions as liaison pilot and public relations officer. Howard E. Brown, Jr., has been named as the California company's District Office supervisor.

P. G. Cobble has been promoted to managing director of Greer Aircraft, a member of the Hervey Safety Group.

Kenneth R. Lennard and Donald Carson have been named manager and assistant manager respectively of the newly formed spare parts division of the Night Avionics-Security Division.

Glen J. Stockhoff has been promoted to head Glens E. Martin Co. a Electronic Property services as well as the company's representation of its Electronic department. Sackoff himself has been placed in charge of the Radio Frequency section and Dr. Fred E. Humes is in charge of the Pulse and Service section.

Joe Crocker has been made manager of public relations, Whitehouse, D. C. for Team World Airlines. Milton Carson has been named director of analysis and reports for the carrier according to Hines McGee, who will make regular business. It will be based in Kansas City.

### Honors and Elections

Dr. Harold E. Dye, Team World Airlines director of medical services, has been elected as president of the Aulin-Vieland Division, Ann.

George S. Chamberlain has been named a director of Schweizer Aircraft Corp., El Paso, N. T. He is a sales engineer for the firm's new products division.

Edward L. Berg, new president-elect of E. S. Airlines, has been elected to the director's board.

## INDUSTRY OBSERVER

New, Ryan built wing tanks for the Boeing B-47B have a capacity of 1,790 gal. each, according to the firm pointed out the tanks in each one half more capacity than the 1,200 gal. winging tanks used on the North American B-45. The B-47 tanks are believed the largest now in use anywhere.

Latest version of Robert Edison Fulkner's Amphibious two-place flying automobile, has a new full carterage was adapted from the wing used on the Goodrich GA-3 amphibious. Other innovations besides a plane on new point job include a new drive transmission to the wheels at the intermediate equipment.

Piper Aircraft Corp. recently took a look at the Helicopter, the first place one five developed by Prof. Otto Koppert of MIT and Prof. Louis Bellinger of Harvard Business School, with a view to possible production, but decided not to try it yet now. American Aircraft Corp., which was in development the plane, dropped out of the picture some time ago. Industry experts say that the plane can be produced tomorrow, in Texas.

Three Marine 1085 helicopter, similar to the Sikorsky S-55 Los Angeles Airways use, have experienced with the new equipment in the field and a general test has been ordered for all military and civilian models of the type. The six includes a series of small changes at the point where tests could be made. The assembly the shift will be designed later to eliminate the slight possibility which these involve.

Boeing Airplane Co. reportedly has more new interest in Stratostars from other countries. About the only way these could be provided would be for Air Force to permit Boeing to divert a few of its C-97 Stratofortress cargo planes for modification to commercial passenger version. Boeing closed out its commercial Stratostars production last about two years ago, after running off 55 planes.

European sources describe the new night-fighter version of the Russian MIG-15 as having a two-place arrangement, with its fuselage lengthened forward of the wings. It has a radome-like structure in forward appearance to that of the F-100 fighter. Wings are modified but not increased in length at the MIG-15 but have rounded tips. Vertical tail is moved forward enough so that makes trailing edge at its tip angle cut instead of further aft, horizontal tail has been lowered about 2 ft. on the vertical fin.

A Russian new version of the MIG-15, with folding wings, and a two seat version for bomber with radial engine are expected for those Soviet aircraft sources now reported to be in service.

Puerto Rico Helicopter Corp. now claims it is the largest rotary wing aircraft manufacturing in the world, space-wise, with approximately 400,000 sq. ft. in its main plant at Maunabo, P.R., plus 152,000 sq. ft. in a leased plant.

Republic Aviation has signed a contract with Finmeccanica of Italy for production of Republic F-84 Thunderbolt spare parts at factories in Naples and southern Italy to supply the planes being used by European nations under the Marshall Defense Assistance Plan. Plans call for the Italian firm to make about 7,000 components of the F-84, which include some 8,000 parts.

Flying Air Chute Co. is installing newly developed stainless steel springs in military parachutes to replace rubber bands formerly used to spring open the chute as the ripcord is pulled. The new arrangement is considered more foolproof in opening chutes at high altitudes and is extremely low frictions.

Navy's use of Panache H-10F and Sikorsky HO4S-1 helicopters for hydrographic survey work in the Caribbean is estimated to have cut as half the time required for individual survey missions.





## Here Are USAF's Sluggers: Convair YB-60 and Boeing YB-52

With retention of flight test programs of the Convair YB-60 (above) and the Boeing YB-52 Stratofortress (below, right), USAF has announced that general outline details of its new proposed heavyweight global sluggers no longer are classified and has released these unclassified news items from various angles.

Reel details of the bombers are also made public, with official approval.

► **YB-60**—Largest all-jet bomber known to be flying: the Convair YB-60 spans 286 ft., some 24 ft. less than its predecessor, the B-36. But it is longer and higher than the B-36, having a length of 173 ft. and height of 60 ft. 9 in. (to top of the radars).

It is powered by eight P6W (57) turbojets producing a total of approximately 50,000 ft. thrust, slung under the

wings in pods which were developed by the Boeing company. Wing sweep was accomplished by using a wedge-shaped structure at the extremity of the center portion of the entire wing. This accomplished putting a roll on the leading edge of the center wing to continue the sweep line to the forelegs.

The plane has an extra landing gear near the forelegs near to protect the tail section.

USAF let the YB-60 contract Mar. 15, 1951, and the first of two airplanes ordered was ready for engine installation only eight months later. The plane flew Apr. 15 at Ft. Worth, 12 days after rollout. No production contract has been provided.

► **YB-52**—Brought a big bomber, although superficially a design progression of the B-47 Stratofortress, is an entirely new

airplane, incorporating lessons learned in design and testing of the smaller Stratofortress, combination of the photos reveals. The B-52A is the production version.

The YB-52 Stratofortress spans 155 ft., is 151 ft. long and height to the top of the cockpit is 48 ft. L.A. the YB-60, it is powered by eight P6W (57) slung under the wings. In the case of the YB-52, this was to prevent a thus, high-speed serial section.

Landing gear is a complex installation attaching into the forelegs, part which going forward and backward which folding in the rear. Small outrigger landing gear near the wing tips maintain balance on the ground.

Study of the design view of the tail (right) reveals that angle of the horizontal surfaces can be varied considerably to provide effective control at very high speeds.



0220









COL. CHARLES A. LINDBERGH, JR., with Lt. Gen. J. H. Cramer at a recent USAP reunion in Greenwich. Lindbergh flew with his wife and five children in New England. He is an unofficial and unpaid USAP advisor.



CLARENCE D. CHAMBERLAIN, 35, poses in rear of the German Panzer sports car he won in Badegry, Czech. He also operates a dairy farm.



REAR ADM. RICHARD E. BYRD studies a globe at his Boston home. Byrd is a Navy consultant on Arctic and polar projects.



RUTH FLIERL, "The American Girl" is now a belly dancer, Calif., formerly. Flierl could lead her flight.



GEORGE W. HALLIEMAN, today a chief, aircraft engineering division, GAN, Washington.



COL. ROBERT BALESTEN is shown sitting at his Pentagon desk. He is U. S. Air Force arch expert.

## Salute to 1927 Heroes

Twenty-five years ago the flyers pictured on this page captured the world's headlines and helped make 1927 one of America's golden years. Here are some of their 1927 exploits:

- Lindbergh flew first solo across the Atlantic, New York-Paris, May 20-21, in Winnie, Spirit of St. Louis.
- Reed attempted New York-Paris flight June 28 with Best Aircraft, Best Airlines and George Noodle in Fokker tri-motor America, crashed on sea near Britain.
- Chamberlain flew New York/Europe, Greenwich, June 4 with Charles Levine in Wright-Bellanca Columbia monoplane.
- Lillie and Hallieman took off from New York City 11 in a Sikorski for Paris, crashed 348 mi. from The Arc.



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## Safer Air Travel Outlook Bright

Guggenheim Center says greatest need today is large investments in ground facilities, airline equipment.

An exhaustive analysis of current aviation safety research studies usually reflect progress and justify an optimistic view for safer air travel of tomorrow, the Guggenheim Aviation Safety Center has announced.

The study shows that the U.S. Census and Coast Bureau together are conducting approximately 900 individual research projects costing more than \$50 million annually to promote aviation safety.

As a result, the number of accidents has been reduced to almost half what it was five years ago, but is higher per passenger mile because today's larger planes carry more passengers.

However, large additional capital investments in safety by airlines, federal governments and manufacturers are called for to take advantage of the greater safety offered by new technical developments. Harry Guggenheim, chairman of the Foundation Council for the Center announced.

► **Investment Needed.** "The large and rapid increase in commercial flying operations has taken place without adequate increase in investment capital. Our environment is dangerous as a whole in operating on major resources, and neither the government nor the airlines have been willing to face the facts. Commercial aviation is like a small boy who has outgrown his clothes. Traffic has the outstripping airborne equipment and ground facilities. Until these have been adequately provided, for stricter government regulation must be imposed," the Foundation chairman said.

Commercial airlines need to spend about half a billion dollars additional on equipment and other hardware of facilities must be created by federal governments and manufacturers to provide adequate ground facilities, Guggenheim declared.

► **Accident Analysis.** Analysis showed that the 1951 scheduled domestic passenger accident rate per million passenger miles was 1.3, compared with 1.1 in 1950 and 1.9 for an average of the three preceding years. The rate is about three times that shown for scheduled passenger traffic, 45 for 1971.

Non-scheduled transport operation for 1950, last available year, showed an accident rate of 3.7. Personnel being showed a decrease in number of accidents and number of fatalities, from 1949 to 1970, but the personnel flying accident rate is still "appalling," the report states, and "unacceptable of consumer improvement."

► **New Projects.** Since the Guggenheim

Center's first report on aviation safety research shortly after its organization a year ago (Aviation Week, Apr. 3, 1971, p. 15), some 274 additional safety research projects have been added to the 400 projects listed last year in conclusions of 60 separate agencies, universities and the aircraft industry.

Research projects in the last year has filled many of the "gaps" in research noted in the first report and additional

work is being undertaken in many of the weaker areas.

- There are supplies of some of the advances in safety work cited.
- Increased use of flight simulation for better training of crew coordination and emergency procedures. Recommendations to make for more opportunities to make accidents for new types of planes more readily available to airlines.
- Fractionalized cockpit control grouping in most efficient locations, based on studies of human capabilities and reactions instead of standardized controls.
- Continuing studies of tolerance to altitude, acceleration and fatigue.
- Better testing of aircrews to en-

## NEWS



## NOTES

### Grumman SA-16's Save Nearly Two Hundred Airmen

Rescue of downed U.S. airmen afloat in the water off Korea are becoming so common that they hardly rate more than a paragraph in the daily wire dispatches from the Far East theatre. The increasingly common nature of these pick-ups tells the rugged conditions under which aviators of the Air Rescue Service have to operate and the terrific bearing which their Grumman SA-16 Albatross amphibious role as well in loading out in open water to reach pilots and aircrews from their worry rooms. Over 170 crewmen have been rescued so far.



Contributing to the rugged saving characteristics of the SA-16 are the wing tip floats built by Edo which not only provide buoyancy in the water, but also carry several hundred gallons of fuel to

give the Albatross long range for rescue missions. The floats also serve as anchoring points for the ski gear when the SA-16, in its amphibious configuration, lands on ice or snow.

Early amphibious components must take the heaving, which floats must endure—especially the wing tip floats of a plane such as the SA-16, which is lighter than a DC-3, in an open water landing. These floats must not only withstand damage but they must remain tight to keep the water out—and the fuel in. Because Edo has specialized for 37 years in the construction of all-aluminum floats, where precision and good workmanship are so important, the industry turns naturally to Edo both for design and volume production of highly built, care all-aluminum components.

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- Philosophy of these safe design principles by the center to be followed by engineers. Design components to fail with safety, to meet rough handling, impact, snag, etc., procedures. The maintenance and operation should be established consistent with average human ability effort and strategies design to protect the aircraft and components against external human error and malfunctions, and to protect occupants in accidents considered survivable.

- Anti-collision flashes and radar equipment.

- Automatic flight recorder developments, are yet aligned by U.S. aviation pending further refinements and pilot acceptance.

- Convertible aircraft developments, sponsored by Army and Air Force, and secondary light control developments sponsored by NACA, to improve approach and landing characteristics and reduce accidents in this area of flight.

- Setting up a program of automated crash landings to be conducted this year by NACA.

- Passenger protection and plane evacuation studies.

- Improvements in structure materials and analytical design, leading to make aircraft safer in every way.

- Airport visibility studies.

- Upper atmosphere weather studies of gusts, turbulence, jet streams, etc.

- Development of improved model aircraft and engine equipment.

### Australia Seeks Air Base Equipment

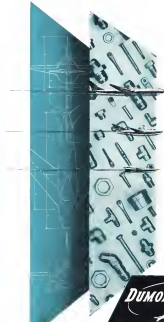
(McGraw-Hill World News)

McBourne—Approval to purchase substantial quantities of foreign-made equipment totally needed in air base construction and equipment has been granted by the Australian Government, with early orders sought.

Orders have already been granted for 12 Lockheed F7V-5 Neptunes, and a like number of DH Vampire T. Mk. 11 two seat trainers. Expected are some English Electric Canberra jet bombers to fill the gap until Australia gets the plant in production. An undoubted number of Gloster Meteor fighters are also due.

The RAAF has placed orders for local manufacture of 55 DH Vampire fighters, 45 Canberras, 72 North American-designed Sabers (with British engines), 24 Vampire trainers and 60 basic trainers of new design (possibly the Commonwealth CA 27).

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## AERONAUTICAL ENGINEERING

### Skyrocket Study Yields Highspeed Data

- Pilot Bridgman's test flights described.
- Transonic speed's effect on subsonic wing told.

Today's transonic market speeds were produced by the extensive program of research aircraft such as the Bell X-1 or the Douglas D-558-II Skyrocket. With supersonic flight speeds only a dozen's throw away, the experience and data gained during highspeed flight tests with research craft have paved the way towards more attainment of super speeds.

Current holder of the altitude and speed records for research craft in the United States is the Douglas Skyrocket with William B. Bridgman up. The craft has been flying for four years, gathering highspeed flight data over a wide range of altitudes and speeds.

That flight experience is worth a close look.

**► Descriptive.**—Although Bridgman calls the Skyrocket an obsolete aircraft, he agreed from a test pilot's point of view that the Skyrocket is the Douglas Skyrocket with William B. Bridgman up. The craft has been flying for four years, gathering highspeed flight data over a wide range of altitudes and speeds.

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SKYROCKET struggles under belly of another ship, Navy F3B 15, before its launch.

2700' rise was experienced due to skin friction, but that the structure absorbed at least half of this and he was never conscious of the lift.

A photostatic recorded all flight instrument readings. Automatically, reading instruments took 400 g's in two measurements on wing and tail surfaces. Control forces and stresses in the structure were measured by means of 900 strain strain gauges and recorded on an oscillograph.

**► Normal Takeoff.**—In the early phases of the Skyrocket program, takeoffs were made in normal manner from the ground. Bridgman reported these as "easy business" because they were made with full rocket fuel load.

Takeoff techniques began with normal starting of the turbojet engine, then jacking of the rocket propellant. After jacking, which takes about 70 sec., boosters were held for 10 sec. while the turbojet was brought up to full power.

Takeoff ground roll began under turbojet thrust alone and showed that over 60 sec., at which time Skyrocket was lifting along the runway at about 100 mph. On the runway, the pilot let the rocket and began actual takeoff. Two revolutions of the propellant were needed to break ground, so will put about 20 ft.

From 100 to 150 mph is just a click of the fingers," said Bridgman. (Over he held the Skyrocket on the ground to 210 mph, throw all the rubber off the tires and took all on the nylon wing.)

Climb was made with rocket gas turbojet power after takeoff, but there were disadvantages. One of these was that Bridgman had to keep an eye on the turbojet instrumentation when he should have been free to watch other vehicles. And the second was that the climb speed was high enough to venturi the turbojet to 17,000 g's when its maximum power was only 11,400.

It took four tries to get an optimum climb in this configuration without encountering the critical Mach number of the engine.

The climb was planned so that about 70 g's of turbojet fuel would be left for the top back home after the test run.

**► Air Landing.**—As the program progressed, everybody realized that the case climb to altitude was punishing the Skyrocket performance. So one of the Skyrockets was converted for pure rocket power and air landing. To do this, the turbojet engine was replaced with additional rocket propellant.

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converted Navy Bearing PDB-15 (USAF B-25) with some work on the aft fuselage added a outlet for the Skyrocket's tail.

The performance of the research craft following the air drop is dependent to a large extent on the mechanics of that drop.

The scientist ship on the crane at release time, and drop took place close enough so that a dead-stick landing could be made on the runway if the rocket failed to function.

Two chase pilots worked with the speed nose. The far flew formation during the drop and checked the firing of all four cylinders of the rocket motor. The second waited at the moment of the speed run (about 100 mi. drop and as high as he could get) and photographed the Skyrocket in its tumbling descent on the way back. This second chase pilot was expected to act as a guide back to landing if, for example, the windshield of the Skyrocket and even, had he also read off the speed to the Skyrocket during final approach and touch down.

Science of the research craft was controlled by the pilot of the mother ship. It was his responsibility to arrive at the exact launch position at proper altitude and speed.

►Skyrocket Flight—Here is a typical drop. Before the drop, the Skyrocket pilot gets aboard. The chase pilot hovers on the harbor side of the mother ship. The rocket powerplant and then the cabin of the research job is processed. The pilot turns on the powerplant.

Right after the drop is made, the first cylinder is fired, followed by the other three in order. The Skyrocket has to be climbed as soon as possible to avoid exceeding critical Mach number, but it takes about 10 sec. before full thrust is available from the engine.

That means there is the possibility of an accelerated stall before climb conditions are attained.

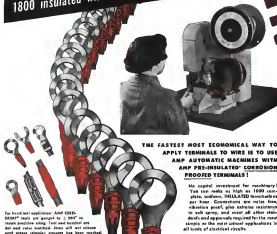
Bradenham also found that he could never pull more than about 1.6 G before the aircraft began to buffet severely. His best stall-warning device was the malfunction of the Skyrocket. Ten hours, which is about all the pilot can see of the engine, whirled violently as load was applied, and Bradenham had it collected accurately at a mild warning.

►Climb and Descent—The Douglas arrangements were never attended with the way Bradenham made the climb with the Skyrocket. Some speed was passed about 1,000 ft. too low and suggested led the programmed speed by at least 0.25 Mach.

But there were things which complicated the climb, and Bradenham. At the rate of fuel consumption—close to one ton per minute—weighing of the craft changes rapidly. So does the angle of attack. Finally the indicated

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angled drops so low that the craft must be allowed to reach supersonic speed angled in climb. When this happens the elevator—which is a conventional subsonic surface—becomes ineffective. Horizontal control at the end of the flight depends on the use of the rocketed stabilizer. This is done with a trim control and does not, of course, operate from the stick.

Pushover at the top of the climb could be controlled adequately and smoothly by changing the bias of the stabilizer. There were no longitudinal problems in the speed runs, but after fuel exhaustion it was necessary to turn 180 deg. to return to base. These turns were started at the highest Mach number reached during the speed runs and were, as Redgreen said, "... very difficult and very educational." (Turning flight will be discussed in detail later in this article.)

►Return and Landing—After the turn had been made toward base, the chase pilot would perform the intercept and then fly formation on the Skyrocket. Another job of the chase pilot was to check the landing gear position of the Skyrocket during the flight back.

Best landing data for glide back was found to be at 360 mph (presumably indicated speed). Returns for return to base were conducted at 15,000 ft. and 360 mph with turn onto base leg at 8,000 or 3,000 ft. Final approach began at 7,000 ft.

Redgreen planned his landings for about one mile down the runway, to have about eight miles left to stop the roll. Touchdown came at about 180 mph.

►Flight Test Program—One of the most unusual phenomena discussed during the test program of the Skyrocket was the development of lateral

oscillations. These oscillations started after the first or two, and appeared on such sensitive flight. What was done by Redgreen and the flight test people to attempt to correct these oscillations under in-flight testing.

►Five Flight-Class Mach number reached 0.91 on the first flight. As the climb passed 35,000 ft., the oscillations became perceptible. The wing dipped about 4 deg. from horizontal at a rate of about 5 deg. per second. (This wing behavior at transonic speed has been reported for several other than the Skyrocket.)

A little later in the climb, rudder oscillations began. The rudder moved about 5 deg. each side of center at about 5 deg. per second. Movement could be stopped by applying rudder pedal force. The lateral oscillations could also be stopped by rudder control, deflecting against the roll. The rudder was found to be much more effective than aileron control.

Pushover at the top of the climb was made gently at about 0.7 G and peak shake was about 41,000 ft.

►Second Flight—Pushover was made at 0.7 G at 47,000 ft. The Skyrocket began to roll during the pushover, and ailerons were ineffective to dampen the motion. Force feedback on the rudder pedals was so strong that Redgreen had to give up thoughts of controlling the plane by those means.

As the roll accelerated, the roll got more violent, and finally Redgreen cut the rudder engine. At maximum Mach number, wing roll was about plus and minus 10 deg., accompanied by about one degree of yaw on each side and terrible rudder travel about 3 deg. each side of neutral. Rate of roll was about 85 deg. per second.

Analysis of this flight and the test



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the rotor assembly was mounted over the cab. A 680-hp. motor engine drove the rotor through its gearbox. The rig has rated up to 1,000 lb. in load.

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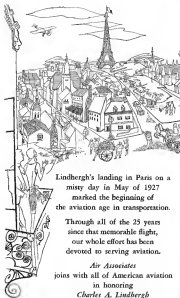
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data suggested looking the rudder for ascending flight. This was done with a plumb which could be withdrawn to free the rudder for landing. Bridge tests over that the rudder lock wasn't expected to be a cure-all, but was intended only to provide enough control to maintain the trim.

► **Third Flight**—Following the installation of the rudder lock, it was found that lateral oscillations still existed, but that adverse effectiveness had returned. This meant controlled flight, and the black weather on this flight was increased. The roll was more sluggish, and Bridgman, and opposite action could be applied to dampen the action.

The daylight data showed that only one-quarter as much adverse force was needed with rudder locked and that aileron angles had been cut to less than half a degree. Rate of roll was down to about 7 deg. per second, and the craft rolled only three rpm and versus 3 deg. In spite of the lock, the rudder was flexing about one-tenth of a degree.

► **Fourth Flight**—On the fifth launch the pushover technique was dropped. The G loading was dropped from about 3.7 to 0.35 in final curve combination after the climb. The result of this flight, Bridgman understands, "was the most interesting of any we encountered."

The Skyrocket began violent rolling as the low load factor was applied. The wings dropped as much as 75 deg. at a rate of the rate of 44 deg. per second. Aileron angles peaked at 44 deg. These oscillations were occurring near the peak maximum Mach number of the craft.

As much as 65-lb. stick force was used in the attempt to try to control lateral motion, effectiveness was so low that Bridgman felt it would be useless to try any higher force. And it was extremely difficult to swing out of phase with the roll, so that adverse motion might improve disturbance.

Finally the flight became so rough that the rockets were cut, but the oscillations continued and became even more violent for a while. The adverse rolling made it impossible to hold the craft in a steady bank to enter the turn to base. Bridgman finally applied a longitudinal load of about 4G in "the opposite direction." "Possibly positive G or a pull-up and the oscillations were damped out enough to make a successful 180-deg. turn."

Reduction of the data showed that at times the rudder hinge moment was as high as 750 in. lb., corresponding to 10-lb. pedal force.

► **Sixth Flight**—In this test—conducted by Bridgman to be the last of the series—the climb angle of attack was increased so rapidly as the buffet at stall warning would allow. Once when the

**seven** out of **ten**  
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boundary was penetrated, the right wing dipped sharply in a partial stall. Momentarily lowering the plane's nose regained control.

Since the top of the cloud there was a small amount of lateral oscillation. The pushover was entered with 0.6G, control was still available, so that was reduced to 0.4G. At this point the left wing got very heavy and could not be pulled up with aileron. So the load factor was increased to 0.6G. This returned lateral control and the load factor was again dropped to 0.6. In a few seconds the propellers ran out and the flight was terminated.

► **Tuning Flight**—One of the most unusual experiences of the flight series was how the turn after speed came. Rudowski says that once a vertical bank and applying full up elevator would not change the aircraft course by one degree. But by using ruddiness turn change of about three or four degrees, and in some cases all that was available, the load factor could be brought to 0G. Then the craft could be put into a diving turn for the top bank to lose.

An undesirable feature of this turn control was that it was mandatory to return the database to neutral position right after the turn started. Electrical activation of the control is slow because of the high air loads on the stabilizers, and to keep the plane from stalling the load factor has to be reduced almost as soon as it is applied.

In one case, the angle of attack reached 18 deg. at the beginning of a turn. The winged was suppressed, but the indicated speed was low. So the turn could be brought to zero, the aircraft pitched forcefully several times about the lateral axis and finally stalled before anything could be done. The emergency informer in this turn came from the speed reduction to subsonic. Cauter of pressure then moved forward about 18 in 1500 which ended the turn by an additional G and



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wheel against the ground!

► **As Landing-Baggage** is on the ground in favor of the air launch technique, and of rocket propellers. He has high praise for the Rocket Motor, but, two-cylinder rocket engines—once started it never failed, he says.

And as for air launch, it "... it was derided, you just drop off into space—you are already flying after a failure—you light off the tubes, burn out your volatile chemicals, give me anything you have left and slide on empty."

Bredemeyer wants a "generally expensive" airplane which would stay at peak Mach numbers for minutes in hand of controls, and which would have proper controls, wings and air control housing. Such a research craft would investigate aerodynamic loading, controls, materials and systems. Could this be a series project of the Douglas X-1?

—EAA

## THRUST & DRAG

The success of a Wright Whirlwind J-1 broke the spirit of duty among us May 26, 1927, at Rome Field. Blue columns plumed off the golden metal covering of a silver Ryan monoplane with "Spauld of St. Louis" emblazoned on the nose. Inside Charles A. Lindbergh made his first check before takeoff.

At 7:52 he opened the throttle, powered down the field and dived the runway to start the greatest adventure aviation had to offer 25 years ago.

His airplane had been built in 68 days for a cost of about \$17,500, including design. Lindbergh and eight St. Louis businessmen had financed the project. The plane weighed about 5,000 lb. and had a 60-hp. monoprop. Wing chord was constant at 7 ft. Power came from a J-5 which developed about 220 hp for takeoff.

Lindbergh's plane was basically a modification of the Ryan M-1 scout and passenger plane. The Ryan of St. Louis had a lengthened fuselage and was cleaned up considerably from the M-1. Structure was simple. Welded steel tubing for the fuselage, laminated spruce spars and leading spar and plywood ribs for the wing.

Visibility for Lindbergh was of a low order; he used a periscope to see forward, and had two side windows and a rear window. He was almost surrounded with pocket doors which were built in tracks between ribs and the engine, the wing center section held a single tank and the outer panels each had one, total fuel load was about 450 gal.

Wing loading of the NK 211 was about 15 p.s.f., and power loading was

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about 22 ft per hr. Wide open, the tool would reach 125 mph.

Today Roosevelt Field echoes to the roar of Thrush and the roaring sounds of turbojets. But 25 years ago it was the birthplace of all Atlantic flight lines, of the myriad daily crossings of Coastliners, DC-6s, Stratojets. Twenty-five years ago America got its first air line of equal status.

A new vibration damper for engine control sticks is being installed on production Miller H-123 and H-125 relay-tow craft. Miller says that it eliminates "jitteriness of the vibration" and that it provides smoothness of control comparable to that of a hand-wing aircraft. It's been tested experimentally and found satisfactory. Miller expects to produce lots for the six to be made by field car chassis on engines which have been cleared. Now all that remains is to eliminate the major source of vibration which will eliminate the need for an additional gadget to stop the stick shake.

What we really ought to be worrying about is a Model T specing, instead of a main hotel, says James H. Wild, one of the original founders of Roosevelt Motors, Inc., and the first American to

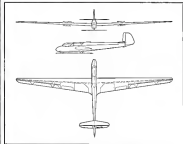
build the now-standard representative aircraft motor.

Wild says that Cessna didn't wait for the construction of the Quonk Elm-beds before setting out on the voyage. And he backs up his statements with some calculations. For example, light aircraft to carry two small men for a couple of days aloft.

Cessna's cockpit is estimated (V) call it nonexistent—in the three-stage engine. Its dimensions are estimated at slightly larger than the V-2 (which was 45 ft long and 5.4 ft in diameter). Cessna and typical engines would be the periphery.

The secret of the amazing performance increase over the V-2 would be in the use of special materials, three-stage construction, increased motor efficiency and reduced weight. Wild estimates the loaded weight as 12.7 tons, empty weight 1.4 tons, reduced later to half a ton by streamlining tanks and engines.

This sounds like the opening gun in the battle between the proponents of the "spiral" theory as envisioned by Cessna's secret Space Flight line, and the "fast things fast" school of Wild and others. Regardless of the outcome of the battle, a conclusion is the fact that the field of space travel has not yet fallen into a single groove of patented thinking. —DMA



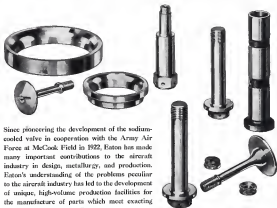
FUCKE-WULF OFFERS 2-PLACE GLIDER

Focke-Wulf GmbH, Bremen, World War II builder of the Fw 239, considered by many to have been the top glider-engine fighter used by the Luftwaffe, is back in business again with a two-place glider named the Korsch III. The dual-purpose glider and high performance glider was scheduled to finish its flight tests and be

placed on the market this month. Construction is conventional fuselage in steel tubing covered with fabric, wing tip of wood, as on the model outside. The Korsch III spans 60.2 ft, length 28 ft, 79.5 ft, wing area including slats 1,217 sq ft. Weight empty is about 300 lb gross, 250 lb. Starting speed is 125 ft/sec.

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## PRODUCTION



RESULTS OF STREAM line high pressure contact point between work and tool, lubricant and work in shear in shear.



MAGNIFIED DRAWING illustrating line of action at the tool tip, showing work chip being removed (clockwise).

## Oil Jet Hits Spot, Stretches Tool Life

- Gulf system shoots the lubricant up at work.
- So chips cannot hinder action at contact point.

Research to boost the efficiency of machining has come up with an answer that promises big dividends for the metal-cutting industry.

Makers of aircraft, engines, accessories and parts, who process unsuitable materials, stress, should benefit greatly by this new approach. It offers greater productivity and lower tool and finishing costs.

The new development is known as High Jet. It was devised by R. J. S. Papp, director of engineering and research, Gulf Research & Development Co. It is a simple but radical method for lubricating and cooling metal cutting tool—long a problem—in the field of machining.

High Jet, Net Flow—Papp's plan reverses the general procedure whereby cutting oil floods the tool and workpiece from above. In the conventional approach, the chip tends to prevent a sufficient volume of oil from reaching the point where the cutting edge meets the workpiece.

Papp's scheme is to use small jets positioned beneath the tool-workpiece contact point. In this way, the stream of cutting oil can penetrate, without obstruction, directly onto the machining point—right where the oil is needed. The result is a new efficiency in hole cutting, coupled with improved heat transfer.



HIGH JET SYSTEM demonstrating how it is applied. Multiple nozzles below work during high-speed stress of oil (lower axis) at cutting edge of tool (top axis).

ITEM	CONVENTIONAL	HIGH JET	GULF RESEARCH & DEVELOPMENT CO.			
			TOOL LIFE	TOOL LIFE	TOOL LIFE	TOOL LIFE
ITEM NO. 1	100% oil	100% oil	100% oil	100% oil	100% oil	100% oil
ITEM NO. 2	100% oil	100% oil	100% oil	100% oil	100% oil	100% oil
ITEM NO. 3	100% oil	100% oil	100% oil	100% oil	100% oil	100% oil
ITEM NO. 4	100% oil	100% oil	100% oil	100% oil	100% oil	100% oil
ITEM NO. 5	100% oil	100% oil	100% oil	100% oil	100% oil	100% oil
ITEM NO. 6	100% oil	100% oil	100% oil	100% oil	100% oil	100% oil
ITEM NO. 7	100% oil	100% oil	100% oil	100% oil	100% oil	100% oil
ITEM NO. 8	100% oil	100% oil	100% oil	100% oil	100% oil	100% oil
ITEM NO. 9	100% oil	100% oil	100% oil	100% oil	100% oil	100% oil
ITEM NO. 10	100% oil	100% oil	100% oil	100% oil	100% oil	100% oil

IMPROVED TOOL LIFE is indicated in these test results obtained at Gulf Research & Development Co., for comparison of overhead flood procedure vs. High Jet.

Translated into practical machining benefits, it means that the operator can use his machine at conventional speeds with greatly extended tool life, as he can push up his operational speed with no reduction in the life of the cutting tool. Gulf reports that at normal cutting speeds life of high-speed tool has



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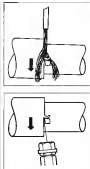
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Overhead fluid (top picture) is applied to chip, jet (below) lets tool contact parts

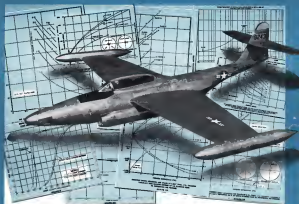
been boosted six to 12 times (even greater boosts are possible at higher speeds). With outside tools, an increased life of at least 5 to 5 times has been obtained. Tied in with the scheme is Galt's Hot Jet oil, a new lubricant selected after extensive research with the spray system.

► **How It Develops** The idea for Piggett's system started when he observed the results of some machining tests using overhead flooding. As the workpiece came away from the cutting tool, he noticed that the work surface was dry and shiny. That meant the flood of cutting oil wasn't having the right spots, so Piggett literally turned things upside down by shooting the oil up to the tool edge for better lubrication and cooling.

The idea was so simple that many an engineer scratched his head and said, "Why didn't I think of it?" The answer to this query, equally simple, was aptly summed up by subjective engineering pioneer Charles Kittinger, who said "It's the absurdity of the obvious."

Once Piggett hit on the idea, the new system was extensively tested in a variety of ways with results ranging from SAE 1023 carbon steel to the new "welder metal" titanium, to prove the advantages of the spray system.

One example reported to illustrate the efficiency of the new method is on the machining of valve stems. Before Hot Jet was installed on the lathe, oper-



Right: A C-124 Globemaster II in the U.S. Air Force's fleet of military cargo planes.

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"Eyes of flight", a new Rohm & Haas film used by the Air Force and the Navy as an official training film on the maintenance of aircraft glazing, is now available for secondary glazing. It is a 35-mil film, contains picture in color and sound. Arrangements for use of the film can be made by writing to the Plexiglas Department, Rohm & Haas Company.

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plan had to change tools after turning out about 50 of the units. With the jet system's improved lubrication and cooling, cutting tools averaged 275 sales runs before they had to be replaced. This meant an estimated saving of about \$1,200 annually per lathe.

►How It Works—Here's how Hilet reports its lubrication and cooling: "Where the tool and the work are in intimate contact, there may be a pressure of 50,000 psi or more, while Hilet's pressure is only a few hundred pounds per square inch. But this doesn't prevent the oil jet from coating the spots that need lubrication and cooling, because these are slight gaps within the both the cutting edge and the work surface that form fine openings.

Even though the jet pressure can't force liquid oil through, the edge jets pressure air will blow the oil's leading point. As the oil hits the edge of the tool a lot of it flashes into vapor and is forced through the minute spaces between the tool and the work.

In the process of vaporization, an insoluble dust is picked up from the cutting edge. Once past the cutting edge, the trapped vapor condenses on the cooler chip and the tool flank and goes off the tool ahead of the cutting edge. There's another way that heat is removed—by an excess flow of liquid oil forced along the underside of the cutting edge.

There's ample evidence that oil is caught between the chip and the tool for an efficient lubrication job. Giff says:

•Chips are coated with oil as they leave the contact area.  
•There's no tool backlap (particles of metal from the work which are welded on to the cutting edge as a result of heat generated).  
•Tool wear rate is reduced.  
•Chip color is more uniform.

Test results have shown a reduction in surface roughness from 200 to 60 microinches, says Giff. Greater dimensional accuracy is another benefit.

In the vaporization process, considerable smoke is generated, but quenchers have been developed to cope with this situation. One of these schemes uses shower heads which pass air from the machine's compressed air system to set up a fine curtain that blankets and condenses the vapor formed.

►Other Benefits—In addition to the direct cooling and lubricating advantages, Giff says that as a result of increased tool life, the setup and grinding time can be reduced on many jobs. Longer tool life also means that fewer tools will be required and the cost prices a real saving on large production jobs.

Widest application for Hilet, says



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Gulf, initially will be on turning operations, because the system can be more easily applied to later tools. But with proper design and attaching fixtures, lift is now practical to many other machining operations.

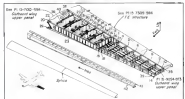
It is reported applicable for use on most other single-point or face tools, milling machines, gear cutters, chucking machines, multiple spindle automatics, some grinding machines, external lathes and certain types of dual speed lathes, reflecting lenses, mounting attachments and tubes or struts are already being designed for some of these uses.

Available Now—Components of the lift system are a motor-driven pump, filter, pressure regulating valve, jet port and holder, and low-pressure coolant spray. Lift jet action side is connected to the cooling coolant system. The discharge is connected by flexible hose to the jet nozzle.

First lift jet production units are slated for delivery early in July. Thompson Products, Inc., Cleveland, at whose plant much of the development work was conducted under production contract, will manufacture them under license from Gulf Research & Development Co. Both Thompson and Gulf's national marketing organization will participate in the distribution of the units.

### Copter Plant Expands

Pasco Helicopter Corp., Mantua, Pa., has started operations in its new 720,000-sq. ft. plant following moving of HUP and H-23 assembly lines into the building. Company now has more than 500,000 sq. ft. in the main plant and in satellite locations. Pasco has awarded James Rodgers, Greenbelt, a contract for a number of HUP maintenance trailers.



### SKETCHES SPEED PRODUCTION

Boring Airplane Co., Seattle, is making liberal use of simplified production sketches detailing airplane assemblies to aid production engineers in assembling components of large bombers and fighters. The sketches eliminate need for hundreds of subminiature blueprints formerly used, save much study

time on subminiature operations. Simple sketch sheet represents Boeing C-97 crew console depicted in photo, clearly useful as simple rigging. Simple system was used in company in World War II to speed B-17 construction and for hundreds of subminiature blueprints formerly used, save much study

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ROLLS on new sheet-forming machine accommodate extended dies with pounds top radius and its underside carrying T-shaped integral stiffeners (shown). Rolls turn on short gut beyond yield point, then end clamping pins provide pull to stabilize sheet for permanent flattening.

### Integral-Stiffened Sheets Flattened

A new process has been devised for getting the required flatness in extended dies with integral stiffeners.

Longmair Aircraft Co., Torrance, Calif., has a machine that is reported to stretch flatten the sheet on a rubber beam, well within the flatness tolerances required.

Details of the type of dies and its potential for stretch wraps and hedges originally were described in *AIRSPACE* News Aug. 28, 1950, p. 15. The sheet is extended in tubular sections with integral stiffeners running from the outside of the tube. For use as dies, these tubular sections have to be split and flattened. But because of internal stresses in the extended section, flatness control has been a troublesome factor in adapting the sheet for structures in high-speed planes.

Initial Temcoing—First, the sheet is run along tubular sections are split longitudinally, then roughly flattened and heat treated. For the metal stretch, the heat-treated sheet is end mounted in a tensioning machine with one V-clamp fixed, while the other is moved by a hydraulic piston. In this setup, the sheet is tensioned to a value which is just below the yield strength of the material.

For the final flattening step, the machine employs a traveling carriage in compressing three displacement rolls. The two lower rolls are moved so that the tops of the extended stiffeners in the sheet rest firmly against the bottom of the rollers, while the top surfaces



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of the rolls contact the sheet as be-  
 tween the stations. The single upper  
 roll, contained over the two lower ones,  
 passes against the internal flat side of  
 the extended sheet.

► **Yield Point Control**—With the re-  
 mable slings automatically maintaining  
 even tension, the traveling carriage pro-  
 gress along the sheet from the end  
 fold in the first slings. After the sheet  
 contacts the last lower roll it angles  
 down under the single upper roll, then  
 angles up again to the second lower roll  
 from which it occurs to the plane of  
 tension between the clamping pins. In  
 this position, the greatest elongation  
 occurs in the material passing under the  
 outer roll. Here, the sheet is tensioned  
 slightly beyond its yield point and work-  
 hardening takes place. As the sheet  
 passes the last displacement roll the re-  
 sidual elongation is a permanently dis-  
 tanced condition.

After this flattening operation, the  
 sheets may be continued, as required,  
 with conventional stretch passes.

#### PRODUCTION BRIEFING

**Bentley Aeroplane Engines** (Bentley  
 Ltd., Montreal, Que.) are plans to build a  
 55-horsepower piston for engine overhaul  
 and repair work. From two a 512-horsepower  
 engine overhaul and repair contract  
 from RCAF.

**Deputy Aircraft Co.**, St. Joseph's,  
 Calif., has agreed a contract with Navy  
 to manufacture the World War II  
 chamber plant in Torrance, Calif.,  
 for use in an aircraft parts fabrication  
 facility. Capacity contains approxi-  
 mately 1 million sq ft of buildings.

**General Dynamics Corp.**, a new offi-  
 cial corporate name for Electric Boat  
 Co., producer of aircraft, submarines  
 and motor and generator.

**Good Central Aircraft Co.**, Tucson,  
 Ariz., has started construction of a new  
 25,000 sq ft. aircraft building, and  
 has scheduled flight camp to cost  
 \$125,000 as part of the \$5 million  
 improvement program being handled  
 by the firm at Tucson.

**Goodrich Aircraft Corp.**, Akron,  
 Ohio, expects to have its new facility  
 for production of rocket components  
 completed this summer.

**Jack & Hinz, Inc.**, Cleveland, has  
 named Aircraft Appliances & Equip-  
 ment Ltd., Toronto, Ont., Canadian  
 distributor for its entire line of automo-  
 tive equipment. The distributor has re-  
 ceived its facilities and its quarters at  
 71 Kipling Ave., Scarb., West Toronto,  
 to handle the new contract.

AIRATION WEEK, May 19, 1952

## Jet Compressor Parts



Robbins Engineering specializes in the production of rotor components and complete rotor assemblies.



Above: Complete rotor assembly for  
 turbojet engine, manufactured by  
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 rotor units require an organization having specialized tooling and  
 inspection equipment and plenty of aircraft know-how. Turbojet  
 engine builders are enthusiastic about the work of the Robbins En-  
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The Robbins organization is cooperating whole-heartedly with the  
 program to build up air power for the defense of our country. All its  
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SPOTTING DANGEROUS storms and terrain is radar's biggest task in active operations, since transport experts feel. That's why.

## ATA Seeks Radar With Fewer Frills

Airlines feel that better terrain and weather warning are essential, navigational features unnecessary.

By Philip Klum

Several important airline radar experts are changing their views on the type of radar the airlines should use. The airlines, they say, need radar primarily for weather and terrain warning purposes, not so for navigation.

"This 'new school' says the airlines don't need all the features and complexity and weight of the military AN/APG-42. But they do think an airline radar should incorporate several new features for better terrain and weather warning, the most drastic of which is the abandonment of the X-band for a new radar wavelength in the 3-6 cm region.

► **Face and Cover**—If this new phosphor persists and the airlines reject the APG-42, it could result in:

- Improved radar screens and terrain warning.
- A lighter, less complex radar designed to airline standards.
- A less costly radar (than the APG-42 or comparable equivalents).

But at the same time it might mean:

- A delay in better radar in the airlines.
- A higher cost program because the airlines would lose the entire cost.

► **Reduced radar methods** for land mapping and as terrain navigation features.

► **A Radar**—This development comes at a surprise to many because most of these airline radar experts on the move men who originally opposed the APG-42 spec for the Navy in 1947. At that time, the APG-42 was expected to be

the ideal transport type of radar device. The development may also come as a disappointment to those who had expected to see airline radar go into use in the near future. The APG-42 is the only U. S. transport radar in production today which comes close to airline needs.

Spokesman for the new group is Frank White of the Air Transport Assn's Air Navigation and Traffic Control division. White was project engineer of the American Airlines group which tested the old APG-10 and later wrote the original APG-42 spec.

► **Why the Change?**—This change in thinking is the just of some of the men who helped write the APG-42 spec in 1947 is not the result of revolutionary new radar developments or techniques. Nor is it an admission of past judgment or lack of foresight in preparing the original spec.

In 1946-47, the following airline business was trying to operate with out needed, over time, precise, accurate, and communications. These pilots who flew radar-equipped planes at that point were very much impressed with its capabilities.

How was a single piece of equipment which could provide ground navigation, warning and storm warning. And if suitably placed radar ground beacons were available, the same device permitted a pilot to fly over desired track and to make instrument approaches.

The uncertainties which surrounded the future of VOR (instrument, ILS (instrument approach), ground-control radar, etc., so that period only served to heighten the enthusiasm of early radar supporters for their new "seeing eye."

It was in the strongest that the spec for the APG-42 was born—spec for a general purpose radar capable of ground mapping and beacon navigation, as well as terrain and storm warning.

► **The Reason**—Today all that has changed. New VORs blanket the U. S. and major airports have ILS facilities. Some of the more than 50 scheduled airlines use an aerial beacon radar (ABR) and precision approach radar (PAR) are already being used.

(Continued on page 55)

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Collins 51R-3 Kibben VOR Receiver

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The winged rooster emblem of Air France, the French National Airline, symbolizes more than world-famous culture and luxury service. It symbolizes, too, a long record of operational and technical achievement — a record recently established by the purchase of Collins 51R-3 receivers to equip the Air France fleet for VOR.

Air France is the first major European airline to fit its fleet with VOR equipment and modern glide slope receivers. In pursuing this great service advancement in Europe, Air France turned to the leader in VOR for equipment — Collins Radio Company. Collins VOR equipment fits with almost every leading airline in the United States, and the proven reputation of this dependable equip-

ment made Collins the natural choice.

Collins 51R-3, 290 channel, VOR navigation receivers will equip the major units in the new Air France installation, with accompanying Collins sensors, sensors and instrumentation providing a space designed for maximum effectiveness and reliability. Utilizing the 51R-3 ILS localizer function, Air France will also install Collins 51V-4 Glide Slope Receivers to complete its installation of modern aids for radio navigation and ILS approach flying.

Collins congratulates Air France on its new VOR/ILS program which is typical of Air France technical leadership in Europe. We are proud that Collins equipment will contribute to this program.

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Actually, users have measured periods between engine overhauls as much as 100% with this great oil!



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Assures superior performance in radial engines. Especially recommended for maximum operating periods between overhauls, it may also be used in horizontally opposed engines when operating conditions do not require a detergent oil.

A fine-quality, non-detergent, straight radial oil, Gulf Aircraft Engine Oil, Series-R, is highly effective in retarding sludge formation. Maintains its body at high operating temperatures, too.

### For More Flying Fun—Don't Settle For Less Than Gulf!

And remember—the Gulf Supply Elementary Gulf Support Dealer locations and many others throughout the U. S. and Canada. Come in, ask us helpfully. Get your copy from your Gulf dealer.

(Continued from page 50)  
Defense measuring equipment (DME) and pictorial computers for better visual flying are on their way.

So the "new school" feels that radar's lead mapping and beacon navigation features are no longer essential requirements for domestic civilian ICAO-compliant Military Air Transport Service and international airlines, which operate in areas without adequate navigational aids nor suit need to sacrifice air transport performance in these roles.

► **New Warning Radar**—The "new school" wants a brand new radar, not a stripped-down version of the APS-41. Here is what they would like in their warning radar:

► **Panel beam only**, instead of APS-41's fan-shaped lead mapping and beacon navigation beam which would be eliminated, to save weight and complexity.

► **No ground beacon interrogation**. This would eliminate the beacon lead oscillation, tuning cavity, and associated extreme frequency control from the radar transmitter.

► **Only one pulse rate** instead of the three pulse repetition rates used in the APS-41. (In the APS-41, one rate is used for short-range precision search, another for weather surveillance and long-range search, the third for beacon operation.) Pulse frequency and wavelength control would be simplified by this step.

► **Single antenna** scan rate instead of two. The slow scan rate which is now used for beacon interrogations would be eliminated.

► **No pitch axis stabilization** and possibly no roll axis stabilization. Roll axis stabilization, a "load" for ground power units, is a desirable feature for a weather and terrain warning radar. But if only pitch stabilization is eliminated, there is little overall saving. Assuming that stabilization is completely eliminated, it would save considerable antenna weight and complexity, as well as two servo amplifiers and a vertical gyro.

► **No target discriminator filter**. First, could no longer get a "magical" look at distant targets on this radar scope.

► **"Radar-to-Radar"** Design—It is possible to design the radar for "Radar-to-Radar" type construction. Airlines who wanted a general-purpose radar would add features by adding "black boxes" to convert the warning radar into a more useful device. A radar user prepared by Lockheed Aircraft for jet transport use typically is based on this principle. "What would apply that idea only to the specialized antennas to permit the addition of stabilization if it is desired."

This design approach would also simplify the problem of meeting individual airline differences over what features the radar should include. The



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6"	6"	1108	30000
8"	8"	1110	30000
10"	10"	1112	30000
12"	12"	1114	30000
14"	14"	1116	30000
16"	16"	1118	30000
18"	18"	1120	30000
20"	20"	1122	30000
22"	22"	1124	30000
24"	24"	1126	30000
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28"	28"	1130	30000
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\*Thermometer (resistance) switches are provided with a safety resistor on all points.

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Manufacturers of Complete Temperature Measuring Systems for Aircraft

### 'Greatest Invention'

Some years ago, two American Airlines (DC-4) and a Pan American Airways (DC-4) were experimentally equipped with modified AFS-100. One AA DC-4 operated for some months on a charter run between Seattle and Anchorage, Alaska.

During this period, AFS Capt. C. A. Lipson, masterfully acquainted with conditions which had caused loss of other DC-4s without incident, immediately observed the experiment, reportedly used his personal "Radar is the greatest invention since the wheel of the auto industry."

An AA report of the DC-4 operation tells of a heavy snowfall and resulting in, Alaska's first which was probably started on the VFR and LP radio communication equipment. "During this time," the report says, "the airborne radar equipment was used. With an low or medium frequency wavelength equipment available other than the radar, it permitted the flight to proceed using the water-hold contact for the radar scope is only cloudy on position."

It is that each construction carefully increases the weight and gain of the basic stepped-down design.

Foley's No. 44 is the need for atmospheric radar has diminished, the demand for warning radar to spot dangerous storms, icing, and terrain has grown.

Warning radar can put and sustain in an aircraft cockpit. By continuously scanning severe turbulence, surface can build passenger confidence and following. Radar may point out potential areas of hail and low wind exposure but damage, although further search on that is necessary.

There is another important element factor. Airline, experimentally equipped with radar, have proven it possible to fly at night and smoothly through storms which had grounded regular airline flights in past times had been out of radar out of their way.

Entry of the Great at transport into service can be made by emphasizing the fact that high end performance may not put great an effort to read a them by being completely around it. What's more, at higher altitudes, there is less time to double-check existing equipment reliability because there is terrain collision. This, plus recent mountain-top accidents, has focused attention on the terrain warning aspect of radar.

McKetter, But-It's (AFS-4) is much improved over the early AFS-100 when it comes to displaying terrain and terrain warning. Its panel beam matches only those areas of danger to the plane. Its higher power (50 kw.) requires single and well, some experts

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\*See S. E. Per 61



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#### ATA/Airline Radar

Detailed Operating Characteristics

(Continued)

Range: 10, 30, 100 and no

Range modern: 3, 6, 10 and no

Maximum power: 10, 30, 100 and no

Power: 10, 30, 100 and no

Power: 10, 30, 100 and no

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only two levels of echo signal, limited evidence to date suggests that without the constant air support, the steep rainfall gradient will cascade farther into the storm center. To add a third air-echo contour level, it is generally agreed, would greatly complicate the scope presentation.

The air-echo contouring feature adds little complexity and weight to radar circuitry. But many feel that it adds much to a radar's usefulness.

► **Storm in Minnesota**—The spherical shape of raindrops gives them a useful characteristic in distinguishing between echoes which could be either a storm or

a mountain. The AFS-42 beam which is horizontally polarized when it leaves the antenna, will be depolarized when it strikes terrain or a solid object. The depolarized echo energy is first to enter the antenna on its return.

However, a circularly polarized beam is vertically polarized when it strikes a standing. Only its plane is changed. This plane shifted, still-polarized energy is almost entirely rejected upon returning to the antenna. Thus, if the radar's normal polarization is changed to circular polarization, it will return a reasonably named echo from terrain, but no echo (or a weak one) from raindrops.

### Saved by Radar

A New RAD now is able today, thanks to its AFS-42 radar, Buckle tells Aviation Week. Doing so is a somewhat approach the central tower operator because confused and give the pilot instructions to turn right, a course which would have placed up the RAD on a mountain. The pilot checked his radar scope, immediately noted the lightning tower, and made the correct turn to the left.

By designing the antenna to permit switching from the normal horizontal polarization to circular polarization, the pilot has an instantaneous discriminator to distinguish between a storm echo and a terrain echo. This feature adds some complexity to the antenna design. It wasn't unexpected in the AFS-42 primarily because it was developed after the production was well along.

Buckle's AFS-42, project engineer, James Goodby, isn't sure that the increased complexity is justified. He believes that the usual long, noisy characteristic of rain echoes is sufficient to identify them at such.

But, as an airline representative put it, if the new airline warning radar is to incorporate air-echo contouring to encourage pilots to seek safe paths through storms, the system seriously would be absolutely certain that the pilot doesn't try to lead a "safe path" through a mountain.

► **Attenuation From Rain**—The attenuation of radar energy by moisture creates three problems for transport radar.

► **Heavy precipitation** of considerable depth results in absorption of the radar beam, so the terrain cannot return a warning echo.

► **The true display** of a dangerous storm may appear considerably shorter on the radar scope than it actually is.

► **It obscures the shape** of air-echo contours. This can cause serious low-gradient areas to look like high gradient (severe turbulence) areas and vice versa.

► **Elow Sauton**—The experts agree that attenuation occurs in precipitation and that when the attenuation is severe, it causes trouble. But they disagree on how frequently these extreme conditions occur and thus how serious the problems are from a practical standpoint.

Frank White agrees that the problem would be all too practical. A transport radar at Omaha recording 20 to 30 mm per hour were seldom encountered. But he points to recent reports by what he considers to be competent authorities. These indicate that the average thunderstorm has a maximum precipitation rate of 45 mm/hr, or higher (previously, rainfall rate of 38 mm/hr has been considered "heavy").

To show the effects of precipitation

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**HOWARD**



BETTER STORM deflection in photo at right is what action was from this time-part motor. Photo of left shows how storm looks on Navy APG-41-type radar.

attenuation, White says that a 40 mm/hr storm 30 mm deep would be deflected in only 4 mm deep on a 3.2 cm (1/2-in.) radar.

P.A. Bagley, Vice-President who takes a long-term view of the problem (part out first if needed) is extremely keen, it doesn't usually occur in an area of such depth. Another optimistic as possible comes from a radar design engineer with a large aircraft manufacturing. He told Aviation Week of his company's studies which indicated that a 50 km X-band radar should penetrate about 11 mm into a rainfall area in which the rate was 10 mm/hr throughout.

If the experts could agree on expected rainfall rates, the problem would be easier to solve. White doubts whether present techniques in use to measure rainfall are as sufficiently accurate for radar attenuation analysis. He points to several recent reports he has for strong feelings on the matter. With such factors as wind speed and rain rate, the experts must investigate further before they can find a well-founded answer. The Navy is anxious and hopes to find out from tests which will be begun soon at their Patuxent, Md., base.

They will run comparative tests using an S-band (1.5 cm) and X-band (3.2 cm) radar. The S-band radar will serve as the benchmark, because it had enough ability only negligible attenuation in rain. The Navy hopes to have some test results by this fall.

The solution—if the weather attenuation problem turns out to be as serious as White and several others think it is, White feels that the most radical radar should move to a new wavelength in the 3-mm to 10-cm region, attenuation goes down as radar wavelength goes up.

At 10-cm, moderate attenuation is negligible. But using this wavelength would mean other problems of antenna size, beam width, etc. The APG-41's 10-in. diameter antenna is the minimum size for carrying transports. Hence

White and others think that a non-practical wavelength of about 6 mm is the answer.

Comparing the 3.2-cm APG-41 with an equivalent 6-cm radar, White says that so heavy rain (10 mm/hr) there would be an 80% decrease in attenuation using the 6-cm set. There would be corresponding improvement at higher rainfall rates.

► The Police—but you never get something for nothing. Changing to a 6-cm set will reduce the radar's beam diameter range. It will also double the beam width to about 10 ft. This gives more vulnerable ground clutter during weather and navigation search operations. It also gives poorer definition for ground mapping.

► Extensive Development—Would a change to 6-cm radar require extensive development of new techniques and components? Frank White doesn't think so. Qualified radar design engineers to whom Aviation Week put the same question agree with White.

The radar design would need to be made in the RF portion of the radar transmitter and receiver. The waveguide plumbing would need to be done in external structures. The antenna feed and rotary joint would need redesign. Also, a new 6-cm oscillator and a 6-cm magnetron would be required. The oscillator is available and a suitable magnetron is believed under military development.

The feeling of radar industry goes next to whom Aviation Week talked to is that a 6-cm radar would require substantial design effort, but no major new developments. However, one engineer quickly added that he thinks that "X-band radar will be found the best compromise wavelength" after the dust settles on the 3-cm vs. 6-cm argument.

If the decision has been made, how soon will they have it in quantity and how much will it cost? How will the decision on which type radar to use affect price, delivery, and possible military targets? Next week's article on this story will discuss these questions.

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## EQUIPMENT

### How Coating Affects Fire Resistance

Coating System	Fill Ratio (%)	Day Weight, (each plus coating, lb.)	Time Required for Failure (min)	Remarks
Uncoated	20	1.5	0.35	
S. F. Goodrich	20	30.3	1	Test coated by S. F. Goodrich Co., Akron, Ohio
Seibond Compound	20	30.5	4	Seibond compound No. HT-2 manufactured by American Latic Products Corp., Los Angeles, Calif.
Alclad (water-based)	20	9.9	4.5	Coated with Alclad paint and over-coated with Amercoat Solution No. 1910.
Uncoated	30	2.3	15	

Tests conducted on standard-Alloy 18-gal. tanks of elliptical shape. SOURCE: CAA Technical Development Report No. 29

### Full Oil Tanks Resist Fires Better

A full oil tank will resist fire better than a partly empty one. And a properly filled steel tank gives better protection than an aluminum alloy one. These are two of the main conclusions reported by CAA's Technical Development and Evaluation Center, Indianapolis, after a series of simulated aircraft emergency engine fires.

The fire tests on aluminum-alloy tanks showed that while a 20% full container collapsed in two minutes, one 75% full stood up for 15 min., perhaps long enough for a plane to land in an emergency.

Tanks made of steel or aluminum alloy, with or without heat-resistant coatings, of circular or elliptical shape. They were in sizes from 1 to 10 gal.

In the course of the tests it was determined that aluminum-alloy emergency tanks and brackets were a design, collapsing rapidly immediately and dropping the still intact tanks.

The most tanks were made of tougher metals for lower tests. Galvanic arc damage and filler caps, and composition heat lines were found to fail before the tests did. Tank shape had no effect on the results.

- **Susceptibility**—Here is a summary of CAA's conclusions:
  - **Aluminum** tanks from engine emergency service: Regardless of material and any flameless fuel tank installed immediately after the engine can be a serious fire hazard.
  - **Steel tanks** are best, if location must be all of engine, steel tanks, with CAA recommended gaskets and filler caps, were still intact after all the oil had leaked away.
  - **Coated, heat-resistant aluminum alloy tanks** with fire level high are not best.
  - **Uncoated aluminum-alloy tanks** with high fuel level are slightly superior to coated ones, but more tanks must of heavier than similar tanks with low level. One such tank 50% full resisted 2,000°F flames for 15 min.
  - **Coated tanks** with 20% oil level lasted up to 45 min., enough longer than an uncoated empty tank to permit the fire-extinguishing system to go to work.
  - **CAA Recommendations**—As a result of the tests, CAA makes these recommendations to industry:
    - **Replace** aluminum emergency brackets and tanks with steel fire resistant materials.
    - **Install** fireproof gaskets in oil pump tanks and filler cap assemblies.
    - **CAA** uses aluminum No. 75 gasketing with screws in the tanks.
    - **Install** spot-light filler caps, good for temperatures up to 2,000°F.
    - **CAA** used a sleeve-type cap, applying a quick-opening type which failed in some early tests.

A heat-resistant treatment that showed up well was Alclad for emergency use. In one test where fuel backup was pushed, an uncoated empty tank lasted 2 min., an Alclad-coated tank 7 min.

Result of the tests was reported by CAA Aircraft Development's J. J. Gorman in Technical Development Report No. 29, "Investigation of the Vulnerability of Aircraft Engines (Oil Tanks in Aircraft Section Fuel)." *Continued*

### Torque Spanner

(McGraw-Hill World News)

London—A new line of torque wrenches for the aircraft industry has been put on the market by Autotek Engineering Co., Ltd., Corp's' Bull's eye, Cardiff, Wales.

Known as the Autotek Mk. 6 Spanner, these wrenches have been tested and approved by the Royal Aircraft Establishment, Farnborough.

The Mk. 6 wrench has several notable features. It is fully automatic, there being no dial or light to watch and no clicks to listen for. The head in which the wrench is set by the operator cannot be moved and it can be used by attaching tubes with pressure gauges, the wrench says.

The tool is unaffected by side loads— it pulled out of square the torque head is mounted. Rapid or slow operation does not affect accuracy, which is maintained to within 1% for over 300,000 consecutive operations at full load.

Operation of the Autotek depends on a cam which positions between two ball races, and has at one end the wide square for attaching to the socket. Running on the cam is a roller housed in a rectifier, which is protected by the body.

The load is applied by a coil spring pressing the roller down on the cam. It is called a bias the handle about the cam, the roller has to force the roller through means of the cam against the pressure of the coil spring and its adjusting nut. The cam is so arranged that once the preset load is applied the wrench becomes free.

The usage of Autotek wrenches varies from a small model of 0.5 lb. ft. to a 50 lb. spanner for 50-250 lb. ft. Single or double handle types are available and anticlockwise models can be provided to special order.

A hydraulic type to generate torque loads of 500 lb. ft. and more is being developed.

The company also produces heat setting rings in various sizes, covering 0-200 lb. ft. and 0-1,000 lb. ft., providing a method of rapidly checking and setting all types of torque wrenches of any make. Test and setting rings outside standard ranges are made to order.



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ELCOTTER (N) draws vapors from wing tank through tube (B) which is connected to tank opening by adapter (C). Section is used by adjusting gap between blower and tube inside (D) which telescopes. Gage (E) tells pressure.

## Tank Contents Purged Faster

The job of purging an aircraft fuel tank of vapors and residual fluid has been cut from hours to minutes by a new vacuum device that can be mounted to the tank opening.

The device should help speed vapor burn and repair work on tanks and can be used for withdrawing other types of fluids in various applications. It was invented by Bernard T. Gross, director of laboratories at Rohr Aircraft Corp., Chula Vista, Calif., who holds patents.

Consisting of a tube arrangement with a plate to fit the tank opening and a conventional blower to suck the vapors from the tank through the tube, this device contains all components in a single frame which can rest on the wing. Depending on tank construction and other factors, the amount of vacuum may be varied by adjusting the gap between tube and blower. This is accomplished by telescoping the tube at the blower end. Section is measured by a pressure-sensing probe.

The unit first removes the vapors already in the tank and by creating a vacuum vapors are removed fluid and draws it out. Essentially it hastens the evaporation process. In other words, when pressure is used, for example, "vapors forced from the tank...are almost immediately replaced by vapors caused by evaporation of the residual fluid... Since the constant evaporation of used fluids takes place rather slowly, a great deal of time may be required..." George R. Hill, Rohr patent analyst, claims.

The section device has proved a

major asset in the Rohr "fill and draw" process of applying sealant coatings to the joints of integral wing fuel tanks. The fill and draw process was developed by Rohr to overcome difficulties in spraying, brushing and clamping.

As Rohr explains, "the vacuum stroke" made it possible to control the thickness of the sealant coating during drawing, because in drawing of the coating by the solvent solvent was eliminated."



Fig 1



Fig 2

## Design Flaw Bolt Was Too Long

Here is what poor design practice can do.

The wing flap actuating torque tube of a certain type of aircraft was connected to a universal joint by a bolt which was slightly too long. An adjacent elevator tube, unsupported for a considerable length, was able to whip (Figure 1).

Result: as flaps were retracted after takeoff, elevation of the plane increased, causing extremely rapid climb. As luck would have it, a service representative of the manufacturer was aboard the aircraft. Seeing the relation between flap retraction and rapid climb, he returned flaps to takeoff position, reducing elevator climb.

Because for jammed elevator is shown in Figure 2.

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## NEW AVIATION PRODUCTS

### Tube Voltmeter

A new vacuum tube voltmeter, reportedly offering greater stability and accuracy than previous types and designed to replace several different instruments by combining their functions, has been developed by Southwest Industrial Electronics Co.

The equipment—the Model R—generally is intended for measuring d.c. volts, from 1 v. full scale to 1000 v. full scale. But it also permits accurate measurement of d.c. millivolts on complete scales, a.c. volts and millivolts, stress and megavolts. Further, it is equipped with a detuning control for making small percentage changes in d.c. voltages and provides for improved accuracy in scaling accuracy.

Power requirements for the Model R are 100-120 v. a.c., 60 c., 50 w. The unit measures 18-1/2x18-1/2x10 in., weighs 34 lb., and is priced at \$625 f.o.b. Southwest Industrial Electronics Co., 2531 Post Oak Rd., Houston 19, Tex.



### Riveting Costs Cut

A novel device for pneumatic rivets used in aircraft work improves quality of riveting, yet speeds production and saves time in building new operators, according to its maker, G. A. Braun, Inc. It is called the Pro-Rivet model Rivetmaster and sells for \$37.50. The model shows the rivet gun to hammer only a preset number of times, then automatically stops or powers to the gun, no matter how long the operator continues to squeeze the trigger or handle.

The 11-oz. device can be installed quickly in the handle of any standard, top-operated pneumatic gun, says Braun, which is it a pistol grip, offset or push-button type. Adjustment of hammer blows to take account of variations in rivet size and other factors is accomplished simply by a calibrated wheel on the unit, locked in place by a spring clip.

An earlier tool has also considerably field testing in aircraft plants and has

been used in limited numbers, Braun says. A prototype of the Rivetmaster originally was developed toward the end of World War II by the Property Co., parent firm of the Braun company. The present model is the first to be scheduled for full-scale production by the company.

G. A. Braun, Inc., Syracuse, N. Y.



### Jet Fires Protected

An improved insulation for protecting jet aircraft has been developed at Chance Vought Aircraft Co. at United Aircraft Corp. It is now being used by CV on the Curtiss Navy tailfin lighter.

The insulation is wrapped around hot protrusions, heat and knock heat thus reducing the engine compartment. The product is a new combination of materials, including Fiberglas and aluminum foil. According to CV, it has a nominal weight by 15 ft. on the Curtiss, costs about 10-15¢ less than old-type coating, and won't soak up oil. It has been tested continuously at temperatures up to 500°F.

The material was developed by E. V. Chetwood, director research engineer for the company. The problem of fire protection is serious since where engine and hydraulic oil may come in contact with hot air lines has caused jet aircraft maintenance problems. Existing insulation had a tendency to absorb oil and cleaning had to be so that instead of protecting, the insulation became a fire hazard, Chetwood says. Also it became heavier than aluminum foil.

Chetwood experienced difficulty at first in developing the new insulating blanket. A major one was that the aluminum foil outer surface was easily punctured. He overcame this by adding a protective coating to the foil and an adhesive to the edges.

Chance Vought Aircraft Co. of United Aircraft Corp., Dallas, Tex.

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406, 408, 410, 412, 414, 416, 418, 420, 422, 424, 426, 428, 430, 432, 434, 436, 438, 440, 442, 444, 446, 448, 450, 452, 454, 456, 458, 460, 462, 464, 466, 468, 470, 472, 474, 476, 478, 480, 482, 484, 486, 488, 490, 492, 494, 496, 498, 500, 502, 504, 506, 508, 510, 512, 514, 516, 518, 520, 522, 524, 526, 528, 530, 532, 534, 536, 538, 540, 542, 544, 546, 548, 550, 552, 554, 556, 558, 560, 562, 564, 566, 568, 570, 572, 574, 576, 578, 580, 582, 584, 586, 588, 590, 592, 594, 596, 598, 600, 602, 604, 606, 608, 610, 612, 614, 616, 618, 620, 622, 624, 626, 628, 630, 632, 634, 636, 638, 640, 642, 644, 646, 648, 650, 652, 654, 656, 658, 660, 662, 664, 666, 668, 670, 672, 674, 676, 678, 680, 682, 684, 686, 688, 690, 692, 694, 696, 698, 700, 702, 704, 706, 708, 710, 712, 714, 716, 718, 720, 722, 724, 726, 728, 730, 732, 734, 736, 738, 740, 742, 744, 746, 748, 750, 752, 754, 756, 758, 760, 762, 764, 766, 768, 770, 772, 774, 776, 778, 780, 782, 784, 786, 788, 790, 792, 794, 796, 798, 800, 802, 804, 806, 808, 810, 812, 814, 816, 818, 820, 822, 824, 826, 828, 830, 832, 834, 836, 838, 840, 842, 844, 846, 848, 850, 852, 854, 856, 858, 860, 862, 864, 866, 868, 870, 872, 874, 876, 878, 880, 882, 884, 886, 888, 890, 892, 894, 896, 898, 900, 902, 904, 906, 908, 910, 912, 914, 916, 918, 920, 922, 924, 926, 928, 930, 932, 934, 936, 938, 940, 942, 944, 946, 948, 950, 952, 954, 956, 958, 960, 962, 964, 966, 968, 970, 972, 974, 976, 978, 980, 982, 984, 986, 988, 990, 992, 994, 996, 998, 1000, 1002, 1004, 1006, 1008, 1010, 1012, 1014, 1016, 1018, 1020, 1022, 1024, 1026, 1028, 1030, 1032, 1034, 1036, 1038, 1040, 1042, 1044, 1046, 1048, 1050, 1052, 1054, 1056, 1058, 1060, 1062, 1064, 1066, 1068, 1070, 1072, 1074, 1076, 1078, 1080, 1082, 1084, 1086, 1088, 1090, 1092, 1094, 1096, 1098, 1100, 1102, 1104, 1106, 1108, 1110, 1112, 1114, 1116, 1118, 1120, 1122, 1124, 1126, 1128, 1130, 1132, 1134, 1136, 1138, 1140, 1142, 1144, 1146, 1148, 1150, 1152, 1154, 1156, 1158, 1160, 1162, 1164, 1166, 1168, 1170, 1172, 1174, 1176, 1178, 1180, 1182, 1184, 1186, 1188, 1190, 1192, 1194, 1196, 1198, 1200, 1202, 1204, 1206, 1208, 1210, 1212, 1214, 1216, 1218, 1220, 1222, 1224, 1226, 1228, 1230, 1232, 1234, 1236, 1238, 1240, 1242, 1244, 1246, 1248, 1250, 1252, 1254, 1256, 1258, 1260, 1262, 1264, 1266, 1268, 1270, 1272, 1274, 1276, 1278, 1280, 1282, 1284, 1286, 1288, 1290, 1292, 1294, 1296, 1298, 1300, 1302, 1304, 1306, 1308, 1310, 1312, 1314, 1316, 1318, 1320, 1322, 1324, 1326, 1328, 1330, 1332, 1334, 1336, 1338, 1340, 1342, 1344, 1346, 1348, 1350, 1352, 1354, 1356, 1358, 1360, 1362, 1364, 1366, 1368, 1370, 1372, 1374, 1376, 1378, 1380, 1382, 1384, 1386, 1388, 1390, 1392, 1394, 1396, 1398, 1400, 1402, 1404, 1406, 1408, 1410, 1412, 1414, 1416, 1418, 1420, 1422, 1424, 1426, 1428, 1430, 1432, 1434, 1436, 1438, 1440, 1442, 1444, 1446, 1448, 1450, 1452, 1454, 1456, 1458, 1460, 1462, 1464, 1466, 1468, 1470, 1472, 1474, 1476, 1478, 1480, 1482, 1484, 1486, 1488, 1490, 1492, 1494, 1496, 1498, 1500, 1502, 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2832, 2834, 2836, 2838, 2840, 2842, 2844, 2846, 2848, 2850, 2852, 2854, 2856, 2858, 2860, 2862, 2864, 2866, 2868, 2870, 2872, 2874, 2876, 2878, 2880, 2882, 2884, 2886, 2888, 2890, 2892, 2894, 2896, 2898, 2900, 2902, 2904, 2906, 2908, 2910, 2912, 2914, 2916, 2918, 2920, 2922, 2924, 2926, 2928, 2930, 2932, 2934, 2936, 2938, 2940, 2942, 2944, 2946, 2948, 2950, 2952, 2954, 2956, 2958, 2960, 2962, 2964, 2966, 2968, 2970, 2972, 2974, 2976, 2978, 2980, 2982, 2984, 2986, 2988, 2990, 2992, 2994, 2996, 2998, 3000, 3002, 3004, 3006, 3008, 3010, 3012, 3014, 3016, 3018, 3020, 3022, 3024, 3026, 3028, 3030, 3032, 3034, 3036, 3038, 3040, 3042, 3044, 3046, 3048, 3050, 3052, 3054, 3056, 3058, 3060, 3062, 3064, 3066, 3068, 3070, 3072, 3074, 3076, 3078, 3080, 3082, 3084, 3086, 3088, 3090, 3092, 3094, 3096, 3098, 3100, 3102, 3104, 3106, 3108, 3110, 3112, 3114, 3116, 3118, 3120, 3122, 3124, 3126, 3128, 3130, 3132, 3134, 3136, 3138, 3140, 3142, 3144, 3146, 3148, 3150, 3152, 3154, 3156, 3158, 3160, 3162, 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\*Fig. 8, 10, 11, 12

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## LETTERS

### Carriers and MIGs

Your article on Nord Aviation Feb. 25 had one statement which interested me: you might see the carrier of certain statements which have already occurred in Korea. At the bottom of p. 26 you make the statement that "If the Navy ever expects to and landers over enemy territory, the carrier's flight will have to be well in advance of the need of to better than the MIG."

In November, 1951, the Navy did just that. At the time I was Commanding Officer of the 1st Fleet, and out of group, consisting of Douglas Skyward, two North American F-86 fighters and German F-86 fighters, in company with planes from the Valley Forge and Philippine Sea groups, made several attacks on the bridges across the Yalu River at Sinuiju on the extreme opposite corner of Korea from the Sea of Japan where the carrier was operating.

These bridges were destroyed by MIG-15 aircraft as well as by carrier aircraft from both sides of the river. As you know, we were unable to apply to the aircraft from the Chinese side of the river or to shoot the MIGs across the river. Our two groups made numerous attacks on these bridges.

As stated previously, in addition to the MIG-15, our own group was attacked by MIG-15s. These were at least three days when we jet fighters, which were removing our first intention, captured MIGs attempting to shoot down both our bombers and jet fighters. When the first attack was in the late afternoon, our jet fighters had shot down three MIGs in various and had probably destroyed several others. It is interesting to note that one MIG was shot down by our carrier jet group.

The remarkable thing is that none of our planes were shot down by the MIG-15. The jet which had this remarkable record against the MIG was the German F-86 jet. It was not only able to protect our bombers, but had a more of three to nothing in comparison with the MIG, and this occurred in December, 1951. It did not mean to state that the F-86 is better than the MIG, it is not, but perhaps less is the answer.

There remains, incidentally, was not widely published and not generally well known. As you know the Sea MIG was shot down by an Air Force pilot on Dec. 8, 1951. The next day the Navy shot down a MIG. This was done by a pilot from the jet group of the Philippine Sea, and on the 10th, two MIGs were shot down by a pilot from the 1st Fleet, and one by a pilot from the Valley Forge. I said previously, in addition several MIGs were damaged in these battles.

As you pointed out in your article, the Navy is making progress in understanding the aircraft, however, but we do not consider that Nord Aviation is satisfied today. We believe we are closer to the final solution of our problem. It has already been said, as I can assure you that the carrier is not the carrier and the carrier is not the carrier.

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Report Requested—Continued



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of MIG aircraft defending the Yalu bridge was a most formidable defense and not likely to be repeated in the near future.

If it is interesting to note that as these battles the MiGs and the covering jet fighters became involved in the dogfights while our dive bombers were attacking the targets. It was most comforting to us to find that once the MiGs were fully occupied with the jet fighters they were unable not only to assist our dive bombers or to stop our attack.

In several instances MAGs feed upon the dirt bombers, but with the last time one Grassman Flycatcher ate one going down, they instantly seemed I do not mean by due to note that the Grassman Flycatcher get outperforms the MAG, but I do say that on the ground it has a three-to-one edge as a three-to-one advantage over the MAG as a combat. (I believe since I left Korea, one Flycatcher got hit down by MAGs.)

These experiences indicate that the coordinated attack of a cause as group is just as effective now against potential opponents as it was during World War II, and I feel certain that the Navy can and does export its "total leadership over enemy territory" to the least and range of sea vessels.

Carl T. U. Sikes, USM  
Washington 25, D. C.  
Office of CNO-Op-90  
Navy Department

### 'The Fun Is Gone'

My subscription has exposed Aviation Week, one of the best sources of my money, too, with aviation.

About 32 years ago I made my first model airplane. Borrowing it from—ten feet or so. To me that was the beginning of a career. Aviation was my game. I hung around the airports when they built them. I learned to fly. And finally I went to school. My diploma reads "Bachelor of Science in Aeronautical Engineering."

Aerobics was fun in those days. An "elephant punch" was my old shed out by the local rowing pond known as "The Airport." I have taught it as many as a dozen employees. Each of us had a salary, and sometimes we got it in full. Other times we took what there was and waited for the rest of it.

Sometimes we made some drawings and built an airplane. Other times we made an airplane and then made some drawings. When the airplane was ready, we took to flying it, and each one made his own record for all tests. Then came work to correct the plane. Several good airplanes were developed that way. Some of them developed into substantial manufacturing companies. Some were bought up by the established large

company. Others found the going too rough and went bankrupt. But through it all there was one man doing a job they loved.

There were plenty of rough times. A number of times, I can recall, all hands turned out to evacuate an explosion, for some reason, but the fire would help to melt the payload. There was one time when the accident of

our company voted my husband, talked him into another two-week credit on my mom. Ed been locked out for being on work in prison on the roof. But I still had a job, I could hold out until some money came.

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American Co. of America, 1201 Elm St., Washington, D. C. 20004, steel, \$11,111.

American Lathrop Electric Corp., 120 N. LaFayette St., Toledo, O. T. 43601, electrical parts, \$11,111.

Barton, William, 1100 W. Main St., Springfield, Ohio, steel, \$11,111.

Boeing West Coast Mfg. Co., 100-110 St. 10, Seattle, Wash., steel, \$11,111.

Boeing Aircraft Corp., Wichita, Kan., \$11,111.

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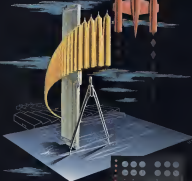
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## FINANCIAL

### Non-Aviation Enterprises Studied

Some stockholders are beginning to ask management what it plans to do if military spending should slump.

Ventures by major aircraft companies into non-aviation fields have once again become a topic of discussion at recent shareholders' meetings.

Such talk is, of course, indicative in the main, since the aircraft industry seems anxious of its cash work in the aviation field as it can handle for the next few years at least. Nevertheless, some stockholders have voiced concern over the plans of these companies in the event of a possible lull.

• **Northrop**—Questions—One such shareholder at the recent Grumman meeting asked just what the company was prepared to do should it suffer the loss of several orders of its military orders. Grumman's president, Louis A. Seifert, replied that the company was exploring possibilities in the non-aviation field where its engineering talents could be profitably applied. This is about as direct an answer as could have been expected, for, under present circumstances, the aviation field is hardly all that bright.

At another shareholders' meeting, that of Consolidated Vultee, Grumman Fleet Orders stated in answer to a shareholder's query that his company, too, was pursuing other projects for development which a slack demand in military order placements. These other projects are in addition to an automotive venture which might result from a failed merger with Kettering and Allen Corp.

For the present, however, the aircraft industry as a whole is rightly and justifiably concerned with the production of aircraft and accessories.

• **Boeing**—Non-aviation enterprises are seeking ways for the industry. For, with the notable exception of Boeing, United Aircraft and Douglas, most companies in the immediate postwar years launched various diversification projects in relative business dried up and disappeared. In 1946 the industry found itself faced with the problem of utilizing war-time production facilities and investing sizable bank balances. The money into other fields was one attempt to find a solution.

• **Boeing**—For example, moved into production of washing machines, small portable engines, general purpose motors, and photographic cameras. At a later date, it acquired a gas appliance plant

manufacturer and insurance company. For one reason or another, Boeing has given up all but the latter two, success here has been pretty much confined to the appliance parts business.

• **Cowles**—which currently seems to have the only active plan to diversify into non-aviation fields, made some of the biggest under previous moves. One of the boldest steps was acquisition of control of ACT Bell Union Co., which produced buses, trucks and truck engines. It also agreed to produce for its former affiliate, Aero Manufacturing Co., kitchen stoves and other appliances. Neither of these developments proved profitable, and in 1948, Cowles had disposed of its ACT Bell holdings and abandoned all non-aviation activities.

• **Boeing**—Boeing which continued to operate a profitable business throughout the difficult postwar period, did so despite its expansion in the manufacture of naval aircraft, trucks and railroads. These operations were all considered a cash money in 1947.

• **Grumman** entered the field of aluminum truck bodies, buses and railroads after the war, with negligible success. The truck body venture proved most profitable, but the rest has never amounted to accounting for more than 4% of revenues. The construction of buses has been dropped and fewer truck bodies are being made, mostly because of the aluminum shortage.

• **Glenn L. Martin** at the end of the war diked into plastics through the formation of a chemical division, and into television broadcasting through cooperation with Westinghouse's television project. Neither of these ventures survived very long. The chemical business was sold to U. S. Rubber and television has been abandoned.

• **Northern Aircraft** (which, as subsidiary, Southern Motors attempted to enter the motor coach field. The company was liquidated in 1948 after a loss of more than \$1 million.

• **Boeing**—Boeing entered the cable into cable manufacture in the postwar period, but this venture, too, was short-lived. After a \$556,000 loss in 1947 the project was discontinued.

• **Successful**—However, not all non-aviation enterprises proved disappointing. One important exception was

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Lockheed's 55 radial engine in Pacific Pioneer Corp. Through totally overhauled to the modern field, the engine has attained a steady profit to Lockheed.

In the same category is Sperry's 1947 purchase of the New Holland Machine Co., manufacturer of farm equipment. This investment has contributed only slightly to Sperry's problem earnings and is continuing to do so.

On the whole, however, an optimistic forecast by the current industry has not met with such success. Thus, of course, they are not aware that future moves along similar lines will follow the same pattern. But the stockholder

naturally must regard the past as indicator of the future and cannot be blamed for regarding the aviation situation with a somewhat skeptical eye. Most observers agree that an aircraft company stockholder expects management to stick to aviation, with the exception of the company depending in a large degree upon the direction of aviation activities, and not diversification.

During the foreseeable future, even more into other than aviation fields by the aircraft industry will remain little more than a non-existent type. The immediate problem of producing aircraft is far more important.

—Selig Altschul



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## AIR TRANSPORT Allison Reports New Turbolineer Progress

- Prop control problems have delayed project.
- But transport now is piling up flight time.



TURBOLINEER: Allison is showing toward 1914 completion of turboprop transport.

Steady headway toward safe, powerful U. S. turboprop transport is disclosed by Allison Division, General Motors Corp., in a progress report reviewing the first 315-hr flight operation of the Allison Turbolineer, first U. S. turboprop carrier.

The airplane, basically a Cessna 240 outfit for its powerplant installation, was purchased from Cessna. The two manufacturers jointly study the initial tests of Allison T35 turboprop engine and Aeromarine propeller in installed assembly. In its new configuration the Turbolineer made its first flight Dec. 20, 1959.

Here are some highlights of recent findings on performance:

- Minimum continuous power 30% greater than the standard power rating of the D3500 engine in conventional Cessna.
- Speed 34 mph faster than the sign for Cessna, at 15,000 ft.
- Fuel flow rate over 2,200 gal. with no increase in gross weight 240 full tank capacity of 1,500 gal.
- Higher performance would result from reducing the nacelle size to take advantage of the engine's smaller frontal area.

Allison's nearest rival in civilian engineering, D. F. Males, received the test results in a paper presented to the SAE National Aerometric Meeting in New York last month.

He supports the conclusion of British Rotaprop Airways chief executive Sir Peter Masefield that maintenance and overhaul should be less costly than for the piston engine, and operating cost also should be less "zero."

- Pistonless—70% of the flight character after which proved new problems, none was of greater consequence as of more serious implications than the problem of windmilling disc," Males says.

So the most vital requirement is for Lockheed automatic feathering. The disc of a turboprop windmilling after power failure in first tests pulled the plane down at 900 ft./min., while the pilot had to hold full throttle and roller. Later, test pilots found that if power were reduced on the good engine,

control was better, but initial descent was 2,000 ft./min., after power was applied to hold speed at 220 ft./min., descent decreased to 100 ft./min.

"From these tests it can be seen that maximum of feathering should be available for emergency operation in case of emergency," Males says.

And quicker feathering is needed for safety in case of a power failure in second, Allison found. To eliminate feathering time and possible loss of control, Allison developed a "dummy pilot" it is a torque-sensing device that decouples the engine from the prop when a torque reversal occurs. This occurs less than a windmilling conventional prop with a stoppage engine.

To speed up changes in blade pitch, hydraulic pressure in the prop system was increased from 5,000 psi to 4,500 psi.

While this gave the needed speed, it was too high for routine maintenance and overhaul in the hydraulic system. So the hydraulic Males was trained to reduce "leaking" valves against which the hydraulic blade pitch control had to work. This allowed reduction of the hydraulic pressure system, and cut bleed-off from 2.15 to 1.15.

Turning Point: As the slow development went on, reliability appeared years off. But Males says a turning point came with tests of the overcompensated propeller control.

The overcompensator is a primitive control to tilt, only if the electronic governor fails. Tests showed that it gave fully stabilized and safe engine operation, although at higher engine speed than normal. This worked even when the electronic control failed at

such a position as to give the wrong signal to the propeller.

After this test, Allison decided to go ahead with planned improvements on the Turbolineer. They put in new pistons, engine, replacing the prototype engine used in the first 50 hr of test (25 flight hr). They put the revised propeller in the place, added a new feathering control on the power section and installed electronic engine meter.

Then followed 10 more flight hr, during which the performance data accumulated show more gathered. With 315 hr of flight test time on the plane, it went through take-off and operation last February.

British Ahead: The British are way ahead of the Turbolineer development. British Turbine Airways plan to start scheduled operations of the Vickers VC-10 into this year. The Vickers has the small Rotolux-Delta-Delta engine, which is simple enough to reduce power involved in prop control and engine meter.

The large-size turboprop Rotolux-Delta-Delta is expected to make its first flight this summer and start scheduled operation in 1964. It has the Rotolux engine, which Corbin Wright is licensed to build.

Meanwhile, neither the Pratt & Whitney T35-powered Lockheed Constellation under New development contract nor the Air France Douglas C-124B with the same engine are expected to be in at least another year.

Which Comes First?—Successful flight testing of the Turbolineer, though not yet near commercial reliability, again poses the question of which will come in scheduled U. S. airline operations







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## Chaos Supreme

Reports coming to Aviation Week from all directions the week following CNA officials are now worried about the magazine's continuing program of public enlightenment on how the Office of Aviation Safety is bungling to the task of investigating and improving air safety. They should be. The CNA's such a loose shooting. Meanwhile, we continue our steady policy of leaving the unsuitability of publicly on credit how CNA's top administration, W. S. Bentley and William Davis, "mismanaged" important technical problems under their direction, although neither has any up-to-date flight or technical or engineering experience. The following lines is from an experienced pilot and technician, within the CNA—A.H.W.

"Recently, there have occurred a considerable number of filter failures in some ILS receivers. When a filter fails the read-out system available to the pilot in the cockpit is unsuitable, i.e., the indicator needle may indicate the aircraft is on course when it is actually off course.

"The procedures of my own available information which making an instrument approach are obvious.

"There are standard check procedures commonly used and when a pilot finds that his ILS receiver is giving him unreliable information he does not make an ILS approach or, in the event he has already started the approach, he discontinues it and pulls up to a safe altitude.

"If this doesn't work, the pilot is to make a low-frequency range approach, CNA approach, ADF approach or to go to some other airport, depending upon the equipment available in his aircraft, weather conditions, and other factors.

"On Apr. 24, 1952, Messrs. Hensley and Davis apparently found out there have been a considerable number of filter failures. No doubt, they applied their "mismanagement" and "suspicious" talents to the discovery and possibly dispatch was hurriedly sent to the field among the thousands from 200 ft ceiling over half mile visibility to 500 ft and final-fourth mile visibility for various using Collins SIR-2 ILS receivers.

"The next to be received, prevention with a capital P. It implies that it is safe to descend to 300 ft with a bad ILS receiver. It suggests compensating for the bad receiver with an additional 100 ft, ceiling and a quarter-mile visibility.

"In actuality, of course, the effect of flying into an off-course obstruction almost inevitably lost high (if the pilot should make an instrument approach with a bad receiver) would be about the same whether the ceiling happened to be 200, 300, 400 or 500 ft at the time.

"Two days later Messrs. Hensley and Davis apparently discovered that some airlines have two SIR-2 ILS receivers installed in their aircraft. So inactive landed dispatches were sent to the field saying that the instrument approach does not apply to these airlines.

"Two days later (Apr. 26), Messrs. Hensley and Davis apparently discovered that at some airports CNA procedure approach rules is installed and operated. So along came another rash dispatch saying that the instrument approach does not apply to these particular cases where CNA instrument is available and not used by the pilot. But that isn't all.

"Still another message was sent out instructing this airport to 'request' the affected airlines to amend their operations specifications to reflect the instrument approach immediately.

"Two days later everybody got another message which said a letter from an airline would be satisfactory in lieu of the second notice to the operations specifications.

"There were also two or three additional messages which were sent out on the same subject—clearly repetitive.

"One of the field inspectors has pointed out to Hensley and Davis that some airlines use the Collins SIR-3 receivers and that these contain the same filter used in the SIR-2 receivers. Apparently, Hensley and Davis have to apply special administrative and executive talent to the formal education because instrument data issued on this date (May 14) still apply only to airlines equipped with SIR-2 receivers.

"It is a pity wonder that the public, the select pilots, the industry and the CNA inspectors have lost faith in the management and direction of CNA? How long will these people be permitted to give instructions, recommendations and suggestions in the name of the U. S. government?

## Engineers Desert Region 3

Another letter to Aviation Week

Alexander McSwery's article, "Engineers Scare in Top CNA Jobs" Apr. 14 was extremely interesting to us, since we are so intimately acquainted with some of the engineers concerned in it.

Your reference to M. B. Hill, Chief, Aircraft Engineering Branch, Region 3, was particularly interesting. The part of it all is that you only touched the surface of the difficulty.

We would like to know why the following Third Region engineering personnel quit CNA, transferred to other regions, or accepted military duty:

- R. J. Auburn, Chief, Powerplant Engineering
- J. C. Bieganski, senior representative
- J. W. Davidson, engineering flight test
- A. A. Gaudin, powerplant engineer
- W. F. Nutter, engineering flight test
- R. E. Schramm, Chief, Engineering Branch
- G. L. Strickland, mechanical engineer
- R. A. Thomas, powerplant engineer

One of these men is wrong? There are none of the men who left but we must, they will be others called M. B. Hill... (or his replacement) practices what CNA engineering training courses teach.

Almost all of the discharges and disaffiliations in CNA would be eliminated if administrative personnel practiced the principles outlined in supervisory courses given at each region by Washington office personnel.

It is a pity that all of CNA's dirty linen must be washed in the public press, but if that is the only way it can be done, then let it be so. CNA is like any other organization, there are a few bad apples that contaminate the whole organization. We are honestly appalled that, by and large, CNA personnel are competent, conscientious and in fact a group of people to work with as will be found in any competent group in industry.

CNA Employees in the Third Region



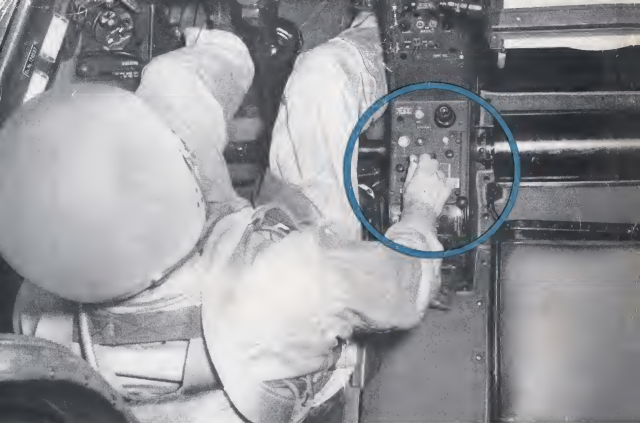
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What's it take to cut a hill down in surmountable? Supplies... men, gear, drops, planes, completely assembled equipment. And getting them there without fuel in the Fairchild C-119's job. Tough terrain can't baffle these battle-proven "Flying Beavers." They deliver in a jangly of a tangled jungle, by parachute or on non-related aircraft.

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ENGINE AND AIRPLANE CORPORATION  
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Other divisions: Engines and Turbine Divisions (Springfield, Mo.) Guided Missile Division, Hartford, Conn.



New G-3 autopilot fits snugly in Douglas F3D-2 fighter cockpit. Device handles all autopilot duties, has additional automatic and control features.

## New All-Electric Autopilot for Navy Jets

**New G-E Autopilot Now in Operational Use in Marine Night-Fighter Squadrons; Other Installations Scheduled**

A new autopilot, the General Electric G-3, is now in operational use in Douglas F3D-2 Skynights and is being installed in Grumman F9F-5P Panthers. It is scheduled for Grumman's new swept-wing F9F-6P, Douglas's A2D Skyshark and other hot Navy planes.

In addition to the main requirements of high speed flight—such as suppressing or eliminating “dutch roll” and high-frequency oscillation—the G-3 incorporates additional control features. It includes continuous automatic synchronization, automatic altitude control, “level-out” and “maneuver-holding” provisions, and allows for wide-angle attitude of autopilot engagement.

All-electric, the G-3 was developed and produced by General Electric in close cooperation with the Navy's Bureau of Aeronautics. It is now in mass production in G-E plants.

The G-3 autopilot is another in the long line of G-E products engineered for the aviation industry by men who know the needs of the air. Whether your needs are for a single generator or a complete bomber defense system—or anything in between—it's a good idea to call your General Electric aviation specialist. *General Electric Company, Schenectady 5, New York.*

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